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CLUSTER SOFTWARE TOOLS

PART I

COORDINATE TRANSFORMATIONS LIBRARY

par

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EUROPEAN SPACE AGENCY

Study of the Cluster Mission
Planning Related Aspects
within the Numerical Simulations Network

CLUSTER SOFTWARE TOOLS

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COORDINATE TRANSFORMATIONS LIBRARY

Version 1.1

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GENERAL REMARKS

This document describes a coordinate transformation library (ROCOTLIB), especially developed in the framework of the CLUSTER mission. Most of the frame of reference used are in geocentric system, and are thus independent of the position of the point of observation. Nevertheless, local frames are also considered. On the other hand, many useful subroutines allowing calendar conversions are also delivered, because time is often given in various format, such as decimal Julian day, or month-day-year, hour-min-sec and so on.

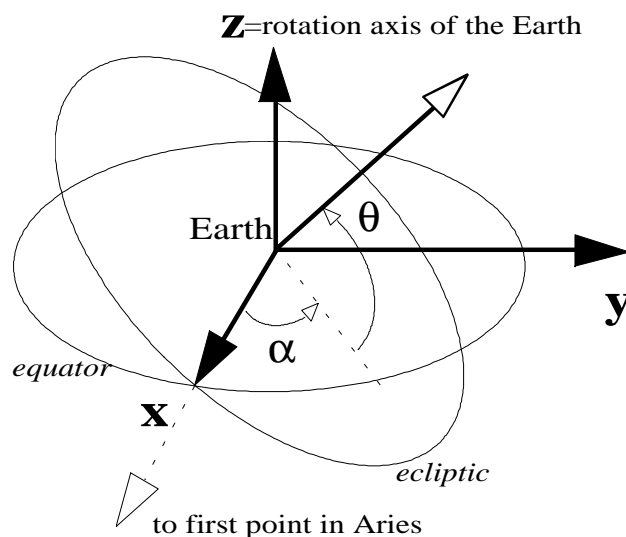
In part I, we recall the definition of the different geophysical coordinate systems used. In part II, a list of transformations is given. Part III gives mathematical expressions of corresponding transformation matrix. Part IV is dedicated to calendar conversion problems. Part V describes the ROCOTLIB developed from the preceding formula, where input/output of each subroutine are detailed. At least, part VI describes directions for use, with examples of Fortran user program. A standard test program, for checking the good installation of the software, is also given with the correct results.

This library is written in FORTRAN 77 language, and can be supported by any computer where this compiler is available.

I- DESCRIPTION OF COORDINATE SYSTEMS

Most of the coordinate systems described are geocentric, with the exception of the dipole meridian system and the VDH system, which are local coordinate systems and thus depend of the position of the point of observation.

1) Geocentric Equatorial Inertial system (GEI)



The Z-axis is parallel to the rotation axis of the Earth.

The X-axis is defined by the intersection of the equator plane and the ecliptic plane, and is pointing towards the first point of Aries (Sun position at the vernal equinox).

one can define the *right ascension* α and the *declination* θ as:

$$\text{right ascension} \quad \alpha = \tan^{-1}(V_y/V_x)$$

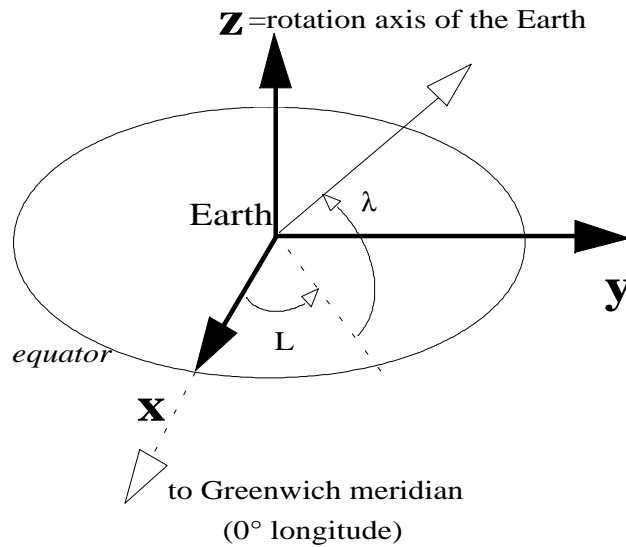
$$\text{with } \alpha \text{ in } [0^\circ, 180^\circ] \text{ for } V_y > 0$$

$$\alpha \text{ in } [180^\circ, 360^\circ] \text{ for } V_y < 0$$

$$\text{declination} \quad \theta = \sin^{-1}(V_z/V)$$

$$\text{with } \theta \text{ in } [-90^\circ, 90^\circ]$$

2) Geographic system (GEO)



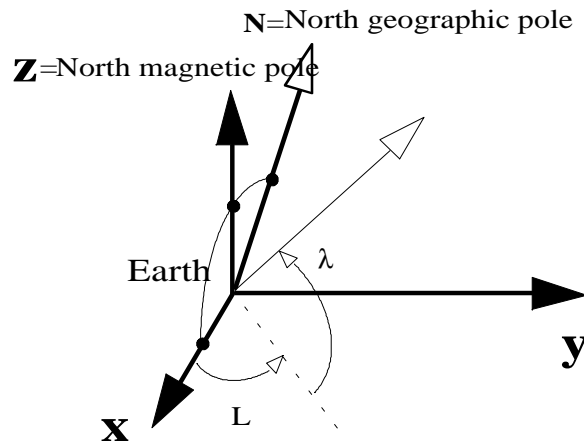
The Z-axis is parallel to the rotation axis of the Earth.

The X and Y axis are included in the equator plane.

The X axis is pointing from the centre of the Earth to the Greenwich meridian (0° longitude).

The GEO system is fixed with the rotating Earth. Longitude L and latitude λ are defined in this system in the same way as right ascension and declination in GEI system.

3) Geomagnetic system (MAG)



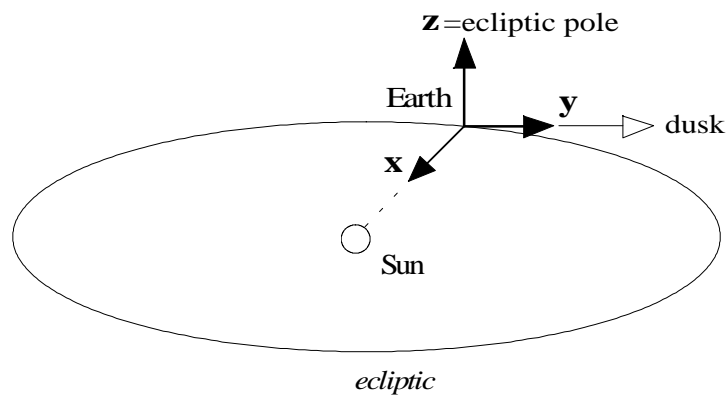
The Z-axis is parallel to the magnetic dipole axis.

If \underline{N} is the North geographic pole, the \underline{N} , \underline{Z} and \underline{X} vector are in the same plane.

The Y-axis is defined as $\underline{Y} = -\underline{Z} \times \underline{N}$

The MAG system is fixed with the rotating Earth. The magnetic longitude L and magnetic latitude λ are defined in this system in the same way as right ascension and declination in GEI system.

4) Geocentric Solar Ecliptic system (GSE)



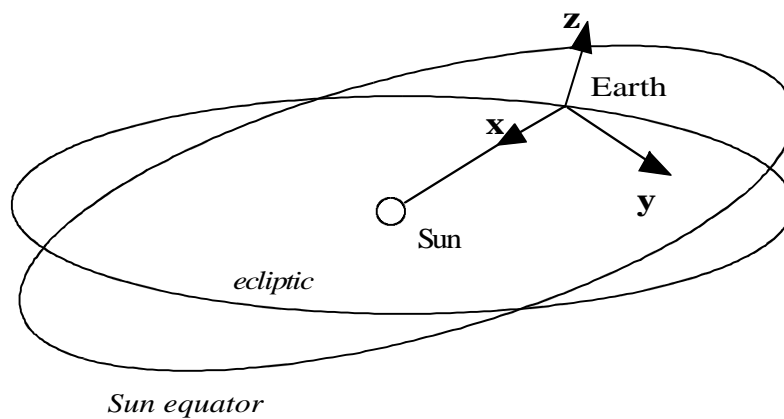
The X-axis is pointing from the Earth towards the Sun.

The X-axis and the Y-axis are include in the ecliptic plane.

The Y-axis is pointing toward the dusk, opposing to the planetary motion.

The Z-axis is parallel to the ecliptic pole. The GSE system has a yearly rotation with respect to the inertial system.

5) Geocentric Solar Equatorial system (GSEQ)



The X-axis is pointing toward the Sun and include in the ecliptic plane.

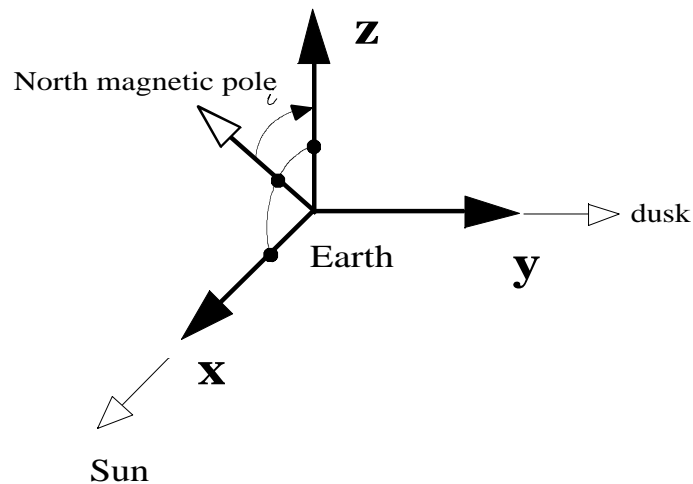
The Y-axis is parallel to the Sun's equatorial plane (inclined to the ecliptic)

X-axis is not necessarily in the Sun's equatorial plane;

Z-axis is not necessarily be parallel to the Sun's axis of rotation (which is perpendicular to Y, and thus in the X-Z plane);

Z-axis is chosen to be in the same sense as the ecliptic pole, i.e. northwards.

6) Geocentric Solar Magnetospheric system (GSM)



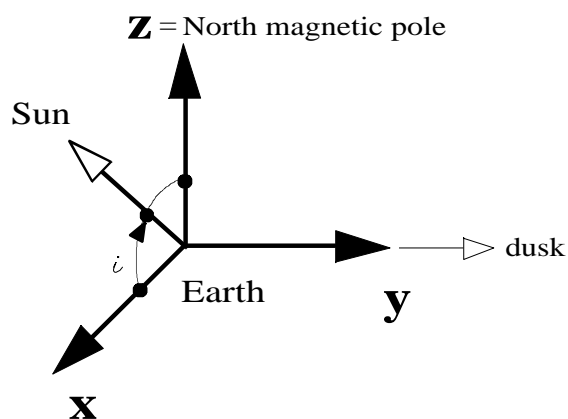
The X-axis is pointing from the Earth towards the Sun.

The X-Z plane contains the dipole axis.

The Y-axis is perpendicular to the Earth's magnetic dipole, towards the dusk and include in the magnetic equator plane.

The positive Z-axis is chosen to be in the same sense as the northern magnetic pole; the dipole tilt angle i is positive when the north magnetic pole is tilted towards the Sun. In addition to a yearly period due to the motion of the Earth about the Sun, the GSM system rocks about the Solar direction with a 24 h period.

7) Solar Magnetic system (SM)



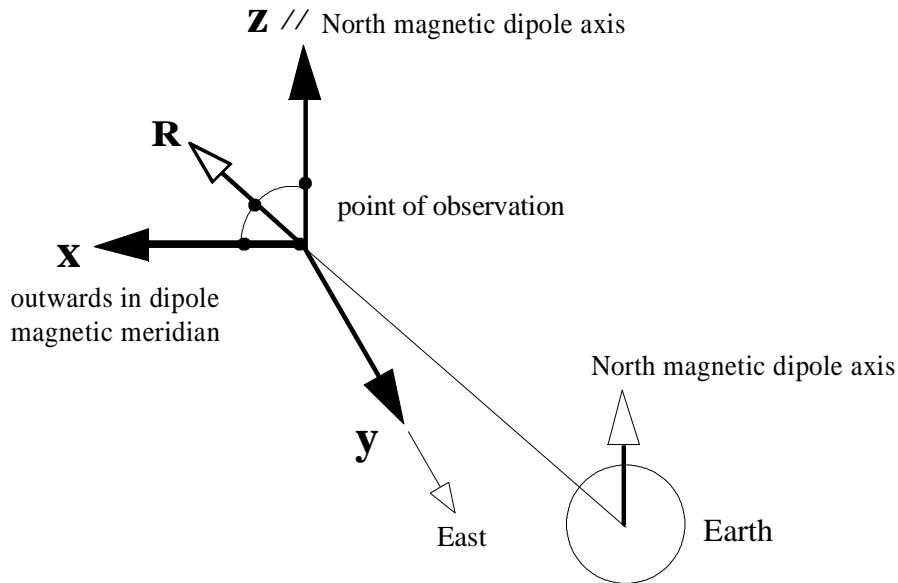
The Z-axis is parallel to the North magnetic dipole.

The X-Z plane contains the direction of the Sun.

The Y-axis is perpendicular to the Earth-Sun line toward dusk.

The SM system rotates with both a yearly and a daily period with respect to the inertial system.

8) Dipole Meridian system (DM)



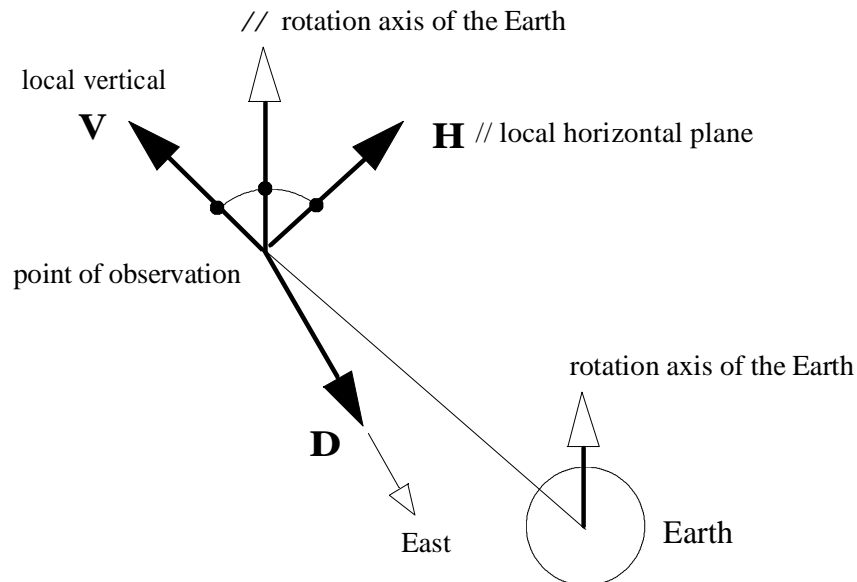
The Z-axis is parallel to the North magnetic dipole axis.

The X-Z plane contains the direction **R** of the point of observation, from the Earth, and is a dipole magnetic meridian plane.

The Y-axis is perpendicular to the **R** vector, eastwards.

This system is a local coordinate system, which is dependent of the position of the point of observation from the Earth.

9) Vertical Dusk Horizontal system (VDH)



The V-axis is the outwards local vertical, to the point of observation.

The H-axis is parallel to the horizontal local plane, positive to the North.

The V-H plane is a geographic meridian plane. The D-axis is azimuthal, eastwards.

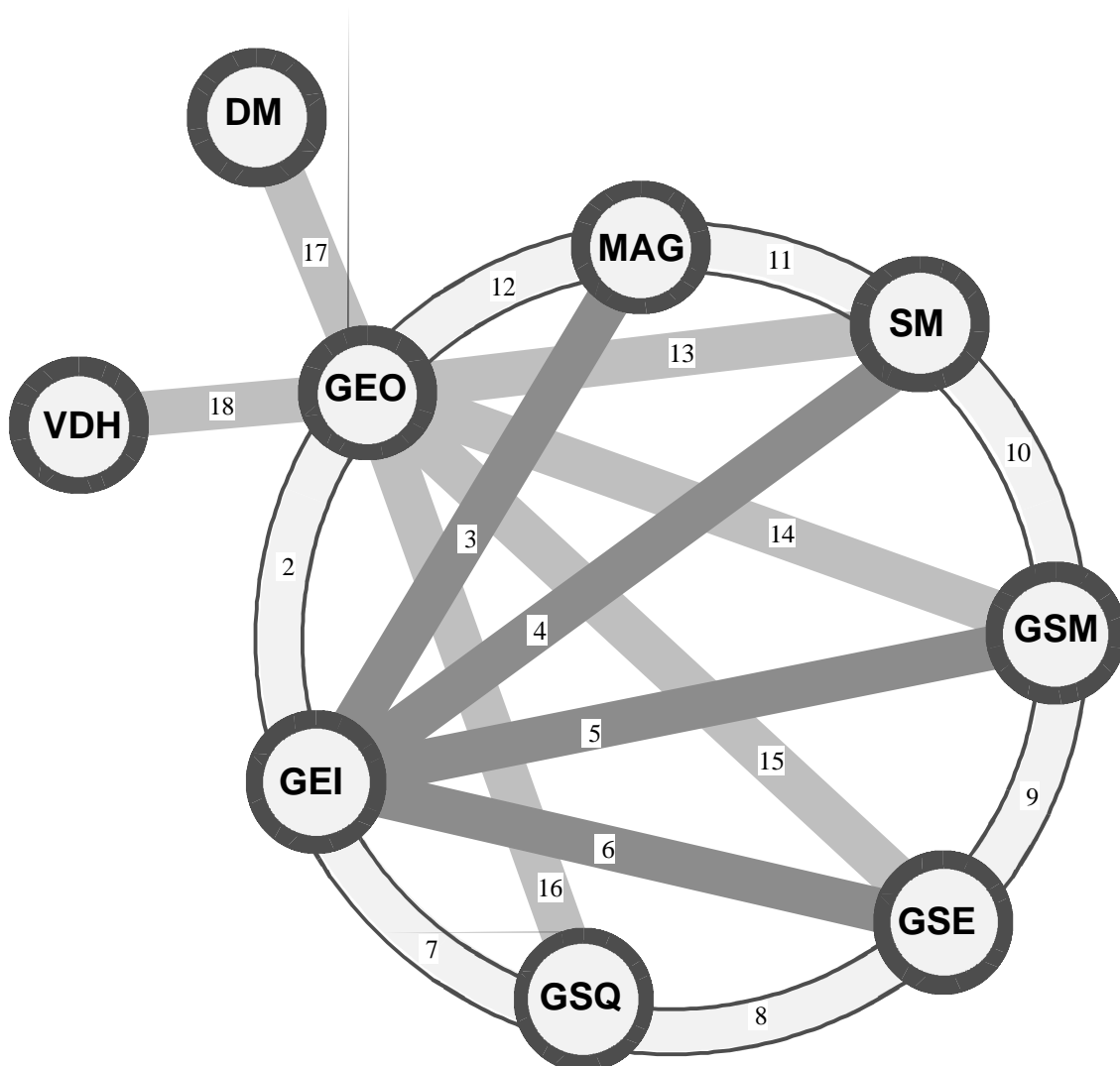
As DM system, this system is a local coordinate system, which is dependent of the position of the point of observation from the Earth.

II- DIAGRAM OF TRANSFORMATIONS

1) general remarks

Among all coordinates systems described in section I, we have single out two particular systems more frequently encountered: the Geographic system (GEO) and the Geocentric Equatorial Inertial system (GEI). From these two system, we have computed all transformations to directly convert these coordinates into any else other. Furthermore, other direct "circular" transformations between the other system are also given, as explained on the schematic diagram below.

2) schematic diagram of transformations



3) list of transformations

All different coordinate transformations are listed below; number correspond to § number of section III, and are mentioned in the above schematic diagram.

	<i>name</i>	<i>input coordinates</i>		<i>output coordinates</i>
2)	geigeo geogei	Geocentric Equatorial Inertial (GEI) Geographical (GEO)	--> -->	Geographical (GEO) Geocentric Equatorial Inertial (GEI)
3)	geimag maggei	Geocentric Equatorial Inertial (GEI) Magnetic dipole (MAG)	--> -->	Magnetic dipole (MAG) Geocentric Equatorial Inertial (GEI)
4)	geism smgei	Geocentric Equatorial Inertial (GEI) Solar Magnetic (SM)	--> -->	Solar Magnetic (SM) Geocentric Equatorial Inertial (GEI)
5)	geigsm gsmgei	Geocentric Equatorial Inertial (GEI) Geocentric Solar Magnetospheric (GSM)	--> -->	Geocentric Solar Magnetospheric (GSM) Geocentric Equatorial Inertial (GEI)
6)	geigse gsegei	Geocentric Equatorial Inertial (GEI) Geocentric Solar Ecliptic (GSE)	--> -->	Geocentric Solar Ecliptic (GSE) Geocentric Equatorial Inertial (GEI)
7)	geigsq gsqgei	Geocentric Equatorial Inertial (GEI) Geocentric Solar Equatorial (GSQ)	--> -->	Geocentric Solar Equatorial (GSQ) Geocentric Equatorial Inertial (GEI)
8)	gsegsq gsqgse	Geocentric Solar Ecliptic (GSE) Geocentric Solar Equatorial (GSQ)	--> -->	Geocentric Solar Equatorial (GSQ) Geocentric Solar Ecliptic (GSE)
9)	gsegsm gsmgse	Geocentric Solar Ecliptic (GSE) Geocentric Solar Magnetospheric (GSM)	--> -->	Geocentric Solar Magnetospheric (GSM) Geocentric Solar Ecliptic (GSE)
10)	gsmsm smgsm	Geocentric Solar Magnetospheric (GSM) Solar Magnetic (SM)	--> -->	Solar Magnetic (SM) Geocentric Solar Magnetospheric (GSM)
11)	smmag magsm	Solar Magnetic (SM) Magnetic dipole (MAG)	--> -->	Magnetic dipole (MAG) Solar Magnetic (SM)
12)	geomag maggeo	Geographical (GEO) Magnetic dipole (MAG)	--> -->	Magnetic dipole (MAG) Geographical (GEO)
13)	geosm smgeo	Geographical (GEO) Solar Magnetic (SM)	--> -->	Solar Magnetic (SM) Geographical (GEO)
14)	geogsm gsmgeo	Geographical (GEO) Geocentric Solar Magnetospheric (GSM)	--> -->	Geocentric Solar Magnetospheric (GSM) Geographical (GEO)
15)	geogse gsegeo	Geographical (GEO) Geocentric Solar Ecliptic (GSE)	--> -->	Geocentric Solar Ecliptic (GSE) Geographical (GEO)
16)	geogsq gsqgeo	Geographical (GEO) Geocentric Solar Equatorial (GSQ)	--> -->	Geocentric Solar Equatorial (GSQ) Geographical (GEO)
17)	geodm dmgeo	Geographical (GEO) Dipole Meridian (DM)	--> -->	Dipole Meridian (DM) Geographical (GEO)
18)	geovdh vdhgeo	Geographical (GEO) Vertical Dusk Horizontal (VDH)	--> -->	Vertical Dusk Horizontal (VDH) Geographical (GEO)

III- MATHEMATICAL EXPRESSIONS OF TRANSFORMATION MATRIX

1) general remarks on transformation matrix

To obtain the matrix to transform a vector expressed in a coordinate system A into another system B, the simplest way is to express the directions of the 3 axis of system B in the coordinate system A.

Indeed, if we notes the coordinates of these 3 unit axes \mathbf{X}_B , \mathbf{Y}_B , \mathbf{Z}_B in coordinate system A as:

$$\mathbf{X}_B = \begin{pmatrix} X_1 \\ X_2 \\ X_3 \end{pmatrix}_{(A)} \quad \mathbf{Y}_B = \begin{pmatrix} Y_1 \\ Y_2 \\ Y_3 \end{pmatrix}_{(A)} \quad \mathbf{Z}_B = \begin{pmatrix} Z_1 \\ Z_2 \\ Z_3 \end{pmatrix}_{(A)}$$

any vector \mathbf{V} can be expressed in system B as:

$$\begin{aligned} V_1^{(B)} &= \mathbf{X}_B \cdot \mathbf{V} = X_1^{(A)}V_1^{(A)} + X_2^{(A)}V_2^{(A)} + X_3^{(A)}V_3^{(A)} \\ V_2^{(B)} &= \mathbf{Y}_B \cdot \mathbf{V} = Y_1^{(A)}V_1^{(A)} + Y_2^{(A)}V_2^{(A)} + Y_3^{(A)}V_3^{(A)} \\ V_3^{(B)} &= \mathbf{Z}_B \cdot \mathbf{V} = Z_1^{(A)}V_1^{(A)} + Z_2^{(A)}V_2^{(A)} + Z_3^{(A)}V_3^{(A)} \end{aligned}$$

Thus the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(B)} = \begin{pmatrix} X_1 & X_2 & X_3 \\ Y_1 & Y_2 & Y_3 \\ Z_1 & Z_2 & Z_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(A)}$$

Similarly the transformation from system B to A is:

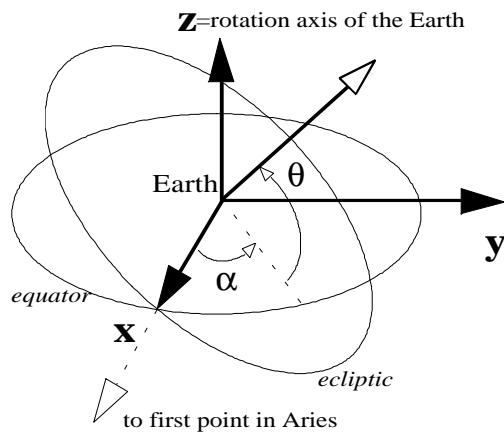
$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(A)} = \begin{pmatrix} X_1 & Y_1 & Z_1 \\ X_2 & Y_2 & Z_2 \\ X_3 & Y_3 & Z_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(B)}$$

All transformation matrix have the following properties, useful for error checking:

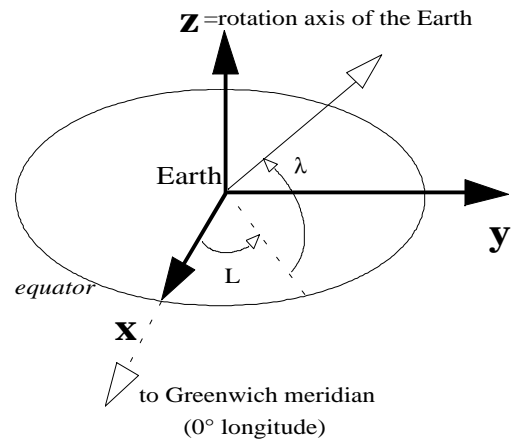
- 1) Each row and column is a unit vector;
- 2) The dot products of any two rows or any two columns is zero;
- 3) The cross product of any two rows or columns equals the third row or column or its negative (Row 1 cross row 2 equals row3; row 2 cross row 1 equal minus row3).

2) GEI to GEO transformations

Geocentric Equatorial Inertial system



Geographic system



GEO and GEI system have their Z-axis in common, so the only difference is a rotation around Z-axis of θ angle, thus the matrix transformation is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} \cos \theta & -\sin \theta & 0 \\ \sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)}$$

and the inverse transformation is obviously:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} \cos \theta & \sin \theta & 0 \\ -\sin \theta & \cos \theta & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)}$$

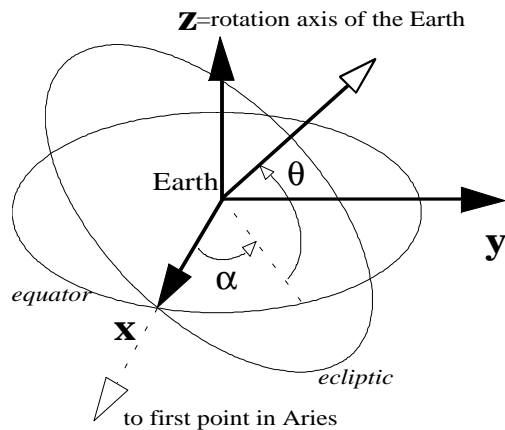
The θ angle is the angle between the Greenwich meridian and the first point in Aries, measured Eastward, in the Earth's equator, from the first point in Aries.

θ is called Greenwich Mean Sidereal Time; GMST is a function of the time of the day and the time of year, since the sidereal day (duration of a day relative to inertial space) is less than 24h.

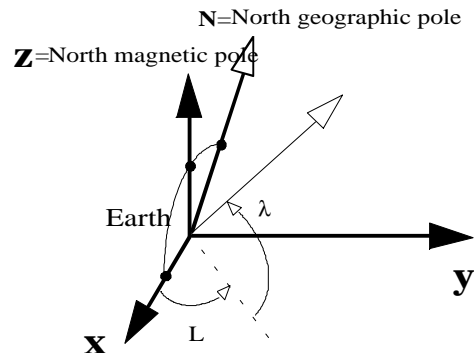
Practically, GMST is computed from CSUNDI subroutine.

3) GEI to MAG transformations

Geocentric Equatorial Inertial



Geomagnetic



Transformation from GEI to MAG system requires a knowledge of the dipole direction in GEI system, noted as **M**.

The geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude, thus:

$$\mathbf{D} = (D1, D2, D3) = (0.06859, -0.18602, 0.98015)$$

Practically, **D** is computed for a given time and year from CDIPDI subroutine.

To know the **M=D** vector in GEI system, which is the Z-axis of MAG system, we use the GEO to GEI transformation computed § III-2, so:

$$\mathbf{Z} = \mathbf{M} = \mathbf{D}_{\text{GEI}} = \begin{pmatrix} D1 \cos\theta & - & D2 \sin\theta \\ D1 \sin\theta & + & D2 \cos\theta \\ & & D3 \end{pmatrix}$$

where θ is the Greenwich Mean Sideral Time computed from CSUNDI subroutine.

We can deduce then the Y-axis of SM system in GEI coordinates from the cross product between North geographic pole (**N**) and North magnetic pole (**M**) :

$$\mathbf{Y} = \mathbf{N} \times \mathbf{M} / |\mathbf{N} \times \mathbf{M}|$$

Normalizing factors occurs because **N** and **M** are not necessarily perpendicular, since **N** has two components equal to zero, cross product is easy and we found:

$$\mathbf{Y} = \begin{pmatrix} -M2 \\ M1 \\ 0 \end{pmatrix} \cdot 1 / (M1^2 + M2^2)^{1/2}$$

X-axis is deduced from:

$$\mathbf{X} = \mathbf{Y} \times \mathbf{M}$$

so:

$$\mathbf{X} = \begin{pmatrix} M1M3 \\ M2M3 \\ -(M1^2 + M2^2) \end{pmatrix} \cdot 1 / (M1^2 + M2^2)^{1/2}$$

All coordinates of X-Y-Z axis of MAG system in GEI coordinates being known, the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} X1 & X2 & X3 \\ Y1 & Y2 & Y3 \\ M1 & M2 & M3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)}$$

Similarly the transformation from system MAG to GEI is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} X1 & Y1 & M1 \\ X2 & Y2 & M2 \\ X3 & Y3 & M3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)}$$

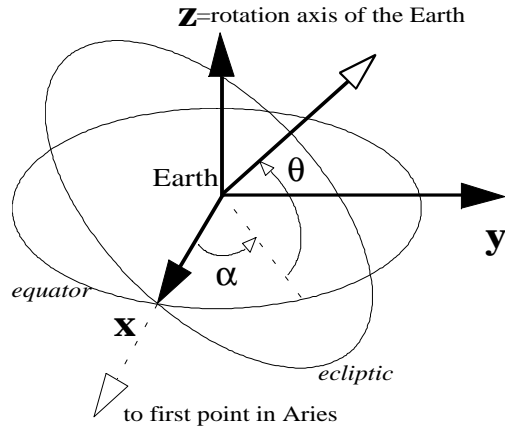
By replacing corresponding values, with $Q = (M1^2 + M2^2)^{1/2}$, we can obtain the fully expanded expression:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} (D1D3\cos\theta - D2D3\sin\theta) / Q & (D1D3\sin\theta + D2D3\cos\theta) / Q & -Q \\ -(D2\cos\theta + D1\sin\theta) / Q & (-D2\sin\theta + D1\cos\theta) / Q & 0 \\ D1\cos\theta - D2\sin\theta & D1\sin\theta + D2\cos\theta & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)}$$

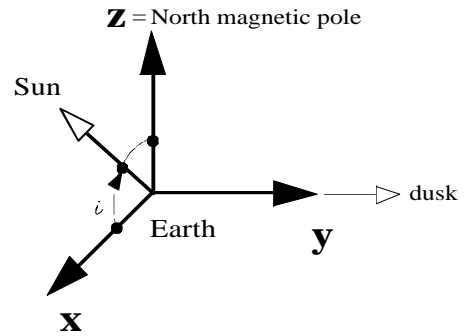
$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} (D1D3\cos\theta - D2D3\sin\theta) / Q & -(D2\cos\theta + D1\sin\theta) / Q & D1\cos\theta - D2\sin\theta \\ (D1D3\sin\theta + D2D3\cos\theta) / Q & (-D2\sin\theta + D1\cos\theta) / Q & D1\sin\theta + D2\cos\theta \\ -Q & 0 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)}$$

4) GEI to SM transformations

Geocentric Equatorial Inertial



Solar Magnetic



Transformation from GEI system to SM system requires a knowledge of Sun direction and magnetic dipole direction in GEI system.

In GEI system, the direction of the Sun is computed from CSUNDI subroutine:

$$\mathbf{S} = (S_1, S_2, S_3)$$

The geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude, thus:

$$\mathbf{D} = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

Practically, D is computed for a given time and year from CDIPDI subroutine.

To know the $\mathbf{M}=\mathbf{D}$ vector in GEI system, which is the Z-axis of SM system, we use the GEO to GEI transformation computed § III-2, so:

$$\mathbf{Z} = \mathbf{M} = \mathbf{D}_{\text{GEI}} = \begin{pmatrix} D_1 \cos \theta & - & D_2 \sin \theta \\ D_1 \sin \theta & + & D_2 \cos \theta \\ & & D_3 \end{pmatrix}$$

where θ is the Greenwich Mean Sideral Time computed from CSUNDI subroutine.

We can deduce then the Y-axis of SM system in GEI coordinates as:

$$\mathbf{Y} = \mathbf{M} \times \mathbf{S} / |\mathbf{M} \times \mathbf{S}|$$

(normalizing factors occurs because \mathbf{M} and \mathbf{S} are not necessarily perpendicular)

X-axis is deduced from:

$$\mathbf{X} = \mathbf{Y} \times \mathbf{M}$$

All coordinates of X-Y-Z axis of SM system in GEI coordinates being known, the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)} = \begin{pmatrix} X1 & X2 & X3 \\ Y1 & Y2 & Y3 \\ Z1 & Z2 & Z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)}$$

Similarly the transformation from system SM to GEI is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} X1 & Y1 & Z1 \\ X2 & Y2 & Z2 \\ X3 & Y3 & Z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

with respectively:

$$\begin{pmatrix} Z1 \\ Z2 \\ Z3 \end{pmatrix} = \begin{pmatrix} M1 \\ M2 \\ M3 \end{pmatrix}$$

$$\begin{pmatrix} Y1 \\ Y2 \\ Y3 \end{pmatrix} = \begin{pmatrix} M2S3 & - & M3S2 \\ M3S1 & - & M1S3 \\ M1S2 & - & M2S1 \end{pmatrix} \cdot 1 / Q$$

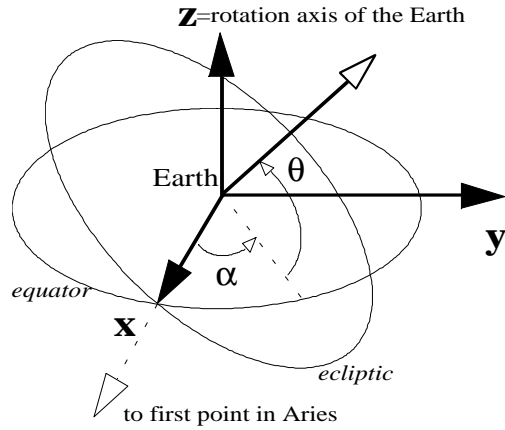
$$\begin{pmatrix} X1 \\ X2 \\ X3 \end{pmatrix} = \begin{pmatrix} Y2M3 & - & Y3M2 \\ Y3M1 & - & Y1M3 \\ Y1M2 & - & Y2M1 \end{pmatrix}$$

$$\begin{pmatrix} M1 \\ M2 \\ M3 \end{pmatrix} = \begin{pmatrix} D1 \cos \theta & - & D2 \sin \theta \\ D1 \sin \theta & + & D2 \cos \theta \\ & & D3 \end{pmatrix}$$

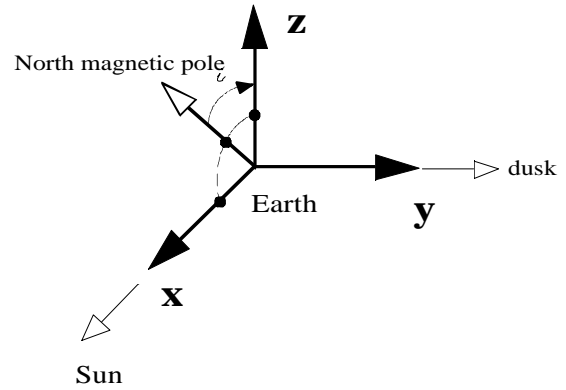
$$Q = \left[(M2S3 - M3S2)^2 + (M3S1 - M1S3)^2 + (M1S2 - M2S1)^2 \right]^{1/2}$$

5) GEI to GSM transformations

Geocentric Equatorial Inertial



Geocentric Solar Magnetospheric



Transformation from GEI system to GSM system requires a knowledge of Sun direction and magnetic dipole direction in GEI system.

In GEI system, the direction of X-axis of GSM system is the direction of the Sun computed from CSUNDI subroutine:

$$\mathbf{X}=\mathbf{S}=(S_1, S_2, S_3)$$

The geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude, thus:

$$\mathbf{D}=(D_1, D_2, D_3)=(0.06859, -0.18602, 0.98015)$$

Practically, \mathbf{D} is computed for a given time and year from CDIPDI subroutine.

To know the $\mathbf{M}=\mathbf{D}$ vector in GEI system, we use the GEO to GEI transformation computed in § III-2, so:

$$\mathbf{M} = \mathbf{D}_{\text{GEI}} = \begin{pmatrix} D_1 \cos\theta & - & D_2 \sin\theta \\ D_1 \sin\theta & + & D_2 \cos\theta \\ & & D_3 \end{pmatrix}$$

where θ is the Greenwich Mean Sideral Time computed from CSUNDI subroutine.

We can deduce then the Y-axis and Z-axis of GSM system in GEI coordinates as:

$$\mathbf{Y} = \mathbf{M} \times \mathbf{S} / |\mathbf{M} \times \mathbf{S}|$$

(normalizing factors occurs because \mathbf{M} and \mathbf{S} are not necessarily perpendicular)

and

$$\mathbf{Z} = \mathbf{S} \times \mathbf{Y}$$

All coordinates of X-Y-Z axis of GSM system in GEI coordinates being known, the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)} = \begin{pmatrix} S1 & S2 & S3 \\ Y1 & Y2 & Y3 \\ Z1 & Z2 & Z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{GEI}$$

Similarly the transformation from system GSM to GEI is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} S1 & Y1 & Z1 \\ S2 & Y2 & Z2 \\ S3 & Y3 & Z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{GSM}$$

with respectively:

$$\begin{pmatrix} Y1 \\ Y2 \\ Y3 \end{pmatrix} = \begin{pmatrix} M2S3 & - & M3S2 \\ M3S1 & - & M1S3 \\ M1S2 & - & M2S1 \end{pmatrix} \cdot 1 / Q$$

$$\begin{pmatrix} Z1 \\ Z2 \\ Z3 \end{pmatrix} = \begin{pmatrix} S2Y3 & - & S3Y2 \\ S3Y1 & - & S1Y3 \\ S1Y2 & - & S2Y1 \end{pmatrix}$$

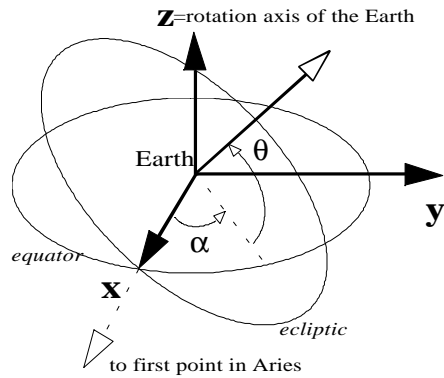
where

$$\begin{pmatrix} M1 \\ M2 \\ M3 \end{pmatrix} = \begin{pmatrix} D1\cos\theta & - & D2\sin\theta \\ D1\sin\theta & + & D2\cos\theta \\ & & D3 \end{pmatrix}$$

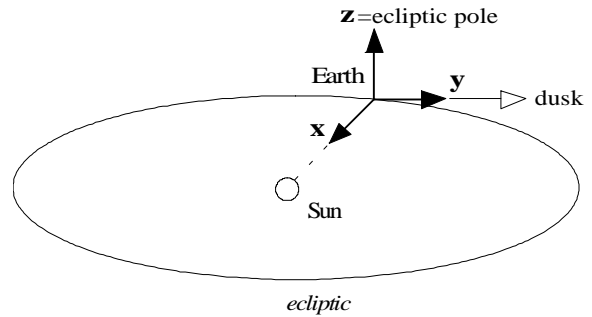
$$Q = \left[(M2S3 - M3S2)^2 + (M3S1 - M1S3)^2 + (M1S2 - M2S1)^2 \right]^{1/2}$$

6) GEI to GSE transformations

Geocentric Equatorial Inertial



Geocentric Solar Ecliptic



In GEI system, the direction of X-axis of GSE system is the direction \mathbf{S} of the SUN , computed from CSUNDI subroutine:

$$\mathbf{X}=\mathbf{S}=(S1, S2, S3)$$

The direction of the Z-axis of GSE is the direction of ecliptic pole, which is a known constant value:

$$\mathbf{Z}=\mathbf{E}=(E1, E2, E3) = (0, -0.398, 0.917)$$

The third axis, Y, is deduced from $\mathbf{Y}=\mathbf{Z} \times \mathbf{X} = \mathbf{E} \times \mathbf{S}$, thus:

$$\mathbf{Y} = \mathbf{E} \times \mathbf{S} = \begin{pmatrix} E2S3 - E3S2 \\ E3S1 - E1S3 \\ E1S2 - E2S1 \end{pmatrix}$$

Thus the transform matrix of any vector \mathbf{V} is:

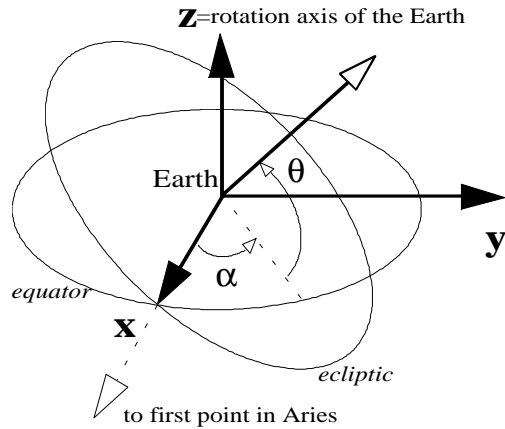
$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} S1 & S2 & S3 \\ Y1 & Y2 & Y3 \\ E1 & E2 & E3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)}$$

Similarly the transformation from system GSE to GEI is:

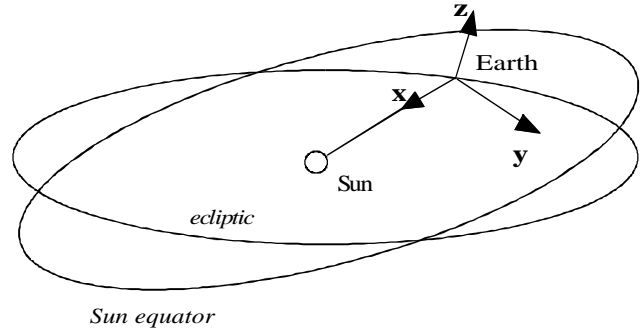
$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} S1 & Y1 & E1 \\ S2 & Y2 & E2 \\ S3 & Y3 & E3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSE)}$$

7) GEI to GSEQ transformations

Geocentric Equatorial Inertial



Geocentric Solar Equatorial



In GEI system, the direction of X-axis of GSEQ system is the direction of the SUN computed from CSUNDI subroutine:

$$\mathbf{X}=\mathbf{S}=(S1, S2, S3)$$

The direction of the rotation axis of the SUN in GEI system is a known constant value:

$$\mathbf{R}=(R1, R2, R3) = (0.122, -0.424, 0.899)$$

Since Y-axis of GSEQ is parallel to the Sun's equatorial plane, the direction of Y-axis in GEI is $\mathbf{R} \times \mathbf{S}$; nevertheless the cross product must be normalized to have a Y unit axis, because \mathbf{R} and \mathbf{S} are not necessarily perpendicular, so:

$$\mathbf{Y} = (\mathbf{R} \times \mathbf{S}) / |\mathbf{R} \times \mathbf{S}|$$

and $\mathbf{Z} = \mathbf{S} \times \mathbf{Y}$

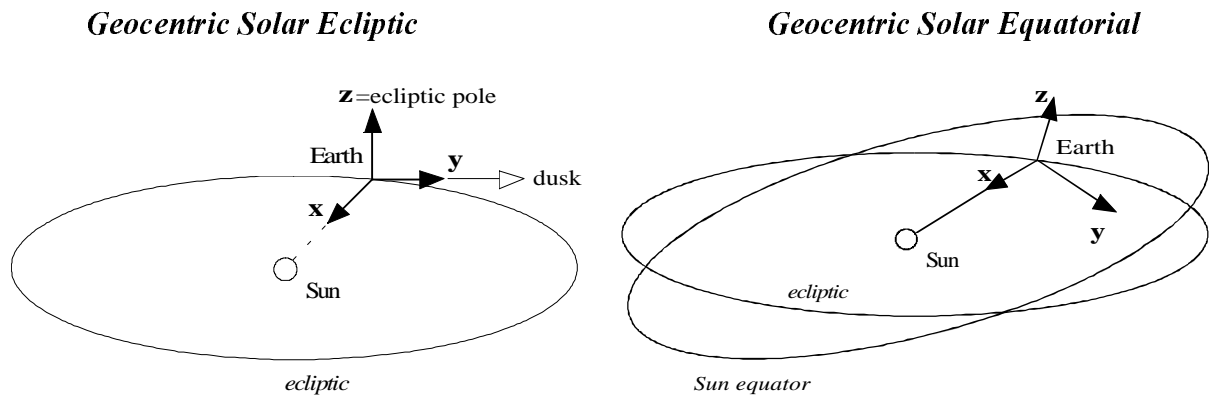
Thus the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSEQ)} = \begin{pmatrix} S1 & S2 & S3 \\ Y1 & Y2 & Y3 \\ Z1 & Z2 & Z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)}$$

Similarly the transformation from system GSEQ to GEI is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} S1 & Y1 & Z1 \\ S2 & Y2 & Z2 \\ S3 & Y3 & Z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSEQ)}$$

8) GSE to GSEQ transformations



The only difference between GSE and GSEQ systems is a rotation about the common X-axis, to have the Y-GSEQ axis parallel to the Sun equator plane.

So, if θ is the rotation angle, the transformation matrix of any vector V is:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GSEQ)} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos \theta & -\sin \theta \\ 0 & \sin \theta & \cos \theta \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GSE)}$$

and the inverse transformation is obviously:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos \theta & \sin \theta \\ 0 & -\sin \theta & \cos \theta \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GSEQ)}$$

computation of θ angle:

θ is the (Y_{GSE}, Y_{GSEQ}) angle, so $\sin \theta = |Y_{GSE} \times Y_{GSEQ}|$

To compute θ , we use the following known vectors in GEI system:

- 1) the direction \mathbf{S} of the SUN , computed from CSUNDI subroutine:

$$\mathbf{S} = (S_1, S_2, S_3)$$

- 2) the direction of ecliptic pole, which is a known constant value:

$$\mathbf{E} = (E_1, E_2, E_3) = (0, -0.398, 0.917)$$

- 3) the direction of the rotation axis of the Sun, which is also a constant value:

$$\mathbf{R} = (R_1, R_2, R_3) = (0.122, -0.424, 0.899)$$

To compute $\sin \theta = |\mathbf{Y}_{GSE} \times \mathbf{Y}_{GSEQ}|$ we use the following properties:

$$\mathbf{Y}_{GSE} = \mathbf{Z}_{GSE} \times \mathbf{X}_{GSE} = \mathbf{E} \times \mathbf{S}$$

and since \mathbf{R} is in the X-Z plane in the GSEQ system:

$$\mathbf{Y}_{GSEQ} = (\mathbf{R} \times \mathbf{S}) / |\mathbf{R} \times \mathbf{S}|$$

so we have $\sin \theta = |(\mathbf{E} \times \mathbf{S}) \times (\mathbf{R} \times \mathbf{S})| / |\mathbf{R} \times \mathbf{S}|$

since $(\mathbf{A} \times \mathbf{B}) \times (\mathbf{C} \times \mathbf{D}) = (\mathbf{A} \times \mathbf{B} \cdot \mathbf{D})\mathbf{C} - (\mathbf{A} \times \mathbf{B} \cdot \mathbf{C})\mathbf{D}$

and $(\mathbf{A} \times \mathbf{B}) \cdot \mathbf{C} = (\mathbf{A} \cdot \mathbf{B}) \times \mathbf{C}$

thus $(\mathbf{E} \times \mathbf{S}) \times (\mathbf{R} \times \mathbf{S}) = (\mathbf{E} \times \mathbf{S} \cdot \mathbf{S})\mathbf{R} - (\mathbf{E} \times \mathbf{S} \cdot \mathbf{R})\mathbf{S} = (\mathbf{R} \times \mathbf{E} \cdot \mathbf{S})\mathbf{S}$

and finally, as \mathbf{S} is a unit vector:

$$\sin \theta = (\mathbf{R} \times \mathbf{E}) \cdot \mathbf{S} / |\mathbf{R} \times \mathbf{S}|$$

For numerical applications, expression

$$\sin \theta = (\mathbf{R} \times \mathbf{E}) \cdot \mathbf{S} / |\mathbf{R} \times \mathbf{S}|$$

can be extended as:

$$\sin \theta = \frac{[(R_2E_3 - R_3E_2)S_1 + (R_3E_1 - R_1E_3)S_2 + (R_1E_2 - R_2E_1)S_3]}{[(R_2S_3 - R_3S_2)^2 + (R_3S_1 - R_1S_3)^2 + (R_1S_2 - R_2S_1)^2]^{1/2}}$$

with numerical values above, this becomes:

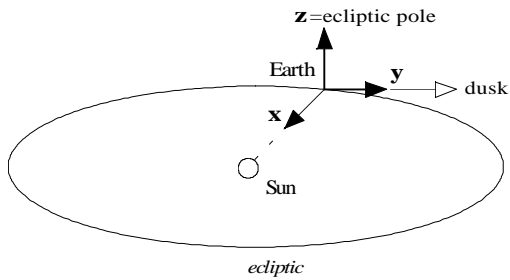
$$\sin \theta = (-0.031, -0.112, -0.049) \cdot \mathbf{S} / |(0.122, -0.424, 0.899) \times \mathbf{S}|$$

Since the Sun's spin axis is inclined 7.25° to the ecliptic, θ changes from -7.25 to 7.25 each year, from approximately December 5 to June 5.

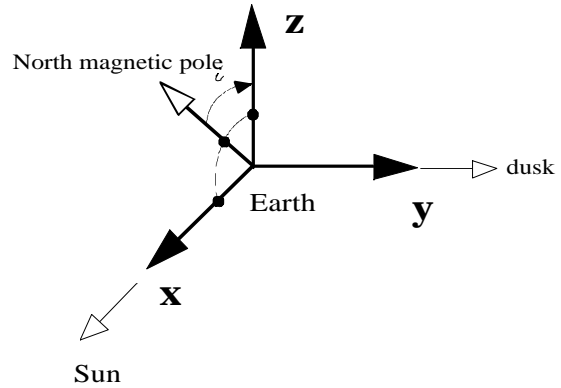
The Sun's spin axis is directed most towards the Earth on approximately September 5 at which time the Earth reaches its northern most heliographic latitude. At this time $\theta=0$.

9) GSE to GSM transformations

Geocentric Solar Ecliptic



Geocentric Solar Magnetospheric



GSE and GSM systems have their X-axis in common, so the only difference is a rotation around the X-axis of the angle ξ , thus the matrix transformation is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos \xi & \sin \xi \\ 0 & -\sin \xi & \cos \xi \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSE)}$$

and the inverse transformation is obviously:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \cos \xi & -\sin \xi \\ 0 & \sin \xi & \cos \xi \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)}$$

Nevertheless the ξ angle cannot be obtained from a simple equation.

To compute the rotation terms of transformation matrix, we use the GSE to GEI and the GEI to GSM previous matrix transformations, given in § III-6 and III-5.

These transformations are noted (see § III-6 and III-5):

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} S 1 & S 2 & S 3 \\ Y 1 & Y 2 & Y 3 \\ E 1 & E 2 & E 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GEI)}$$

with $\mathbf{Y} = (\mathbf{E} \times \mathbf{S})$ in GEI system

and:

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GEI)} = \begin{pmatrix} S 1 & Y 1 & Z 1 \\ S 2 & Y 2 & Z 2 \\ S 3 & Y 3 & Z 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSM)}$$

with $\mathbf{Y} = (\mathbf{M} \times \mathbf{S})$ in GEI system

and $\mathbf{Z} = (\mathbf{S} \times \mathbf{Y})$ in GEI system

we can write the GSM to GSE transformation as:

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} S 1 & S 2 & S 3 \\ Y 1 & Y 2 & Y 3 \\ E 1 & E 2 & E 3 \end{pmatrix} \begin{pmatrix} S 1 & Y 1 & Z 1 \\ S 2 & Y 2 & Z 2 \\ S 3 & Y 3 & Z 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSM)}$$

which give

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} \mathbf{S} \cdot \mathbf{S} & \mathbf{S} \cdot \mathbf{Y} & \mathbf{S} \cdot \mathbf{Z} \\ \mathbf{Y} \cdot \mathbf{S} & \mathbf{Y} \cdot \mathbf{Y} & \mathbf{Y} \cdot \mathbf{Z} \\ \mathbf{E} \cdot \mathbf{S} & \mathbf{E} \cdot \mathbf{Y} & \mathbf{E} \cdot \mathbf{Z} \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSM)}$$

which becomes, since \mathbf{S} and \mathbf{Y} are unit vectors perpendicular between us, as \mathbf{S} and \mathbf{Z} :

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} 1 & 0 & 0 \\ 0 & \mathbf{Y} \cdot \mathbf{Y} & \mathbf{Y} \cdot \mathbf{Z} \\ 0 & \mathbf{E} \cdot \mathbf{Y} & \mathbf{E} \cdot \mathbf{Z} \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSM)}$$

Of course the final matrix does not depend on the \mathbf{S} vector.

Computation of $\cos \xi$

We have to equalize $\mathbf{y} \cdot \mathbf{Y}$ and $\mathbf{E} \cdot \mathbf{Z}$ terms as $\cos \xi$.

Taking $\mathbf{y} \cdot \mathbf{Y} = (\mathbf{E} \times \mathbf{S}) \cdot (\mathbf{M} \times \mathbf{S})$

we compute $\mathbf{E} \cdot \mathbf{Z} = \mathbf{E} \cdot (\mathbf{S} \times \mathbf{Y}) = \mathbf{E} \cdot [(\mathbf{S} \times (\mathbf{M} \times \mathbf{S}))]$

since $\mathbf{A} \cdot (\mathbf{B} \times \mathbf{C}) = (\mathbf{A} \times \mathbf{B}) \cdot \mathbf{C}$

we effectively found $\mathbf{E} \cdot \mathbf{Z} = (\mathbf{E} \times \mathbf{S}) \cdot (\mathbf{M} \times \mathbf{S}) = \mathbf{y} \cdot \mathbf{Y}$

then:

$$\mathbf{y} \cdot \mathbf{Y} = \mathbf{E} \cdot \mathbf{Z} = \cos \xi$$

by replacing the corresponding values, we set:

$$\cos \xi = (\mathbf{E} \times \mathbf{S}) \cdot (\mathbf{M} \times \mathbf{S})$$

since $\mathbf{A} \cdot (\mathbf{B} \times \mathbf{C}) = (\mathbf{A} \times \mathbf{B}) \cdot \mathbf{C}$

we found $\cos \xi = -(\mathbf{E} \times \mathbf{S}) \cdot (\mathbf{S} \times \mathbf{M}) = -(\mathbf{E} \times \mathbf{S}) \times \mathbf{S}$

and finally

$$\cos \xi = \mathbf{E} \cdot \mathbf{M}$$

\mathbf{E} and \mathbf{M} are known since:

1) the direction of ecliptic pole in GEI system is a known constant value:

$$\mathbf{E} = (E_1, E_2, E_3) = (0, -0.398, 0.917)$$

2) \mathbf{M} is the dipole direction in GEI system, computed § III-5 as:

$$\mathbf{M} = (M_1, M_2, M_3) = ((D_1 \cos \theta - D_2 \sin \theta), (D_1 \sin \theta + D_2 \cos \theta), D_3)$$

3) the geographic coordinates of the dipole axis \mathbf{D} is computed for a given time and year from CDIPDI subroutine; for instance for IGRF epoch 1965 we have:

$$\mathbf{D} = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

4) the θ Greenwich Mean Sidereal Time is computed for a given time and year from CSUNDI subroutine.

This leads to the final expanded formula:

$$\cos \xi = E_1(D_1 \cos \theta - D_2 \sin \theta) + E_2(D_1 \sin \theta + D_2 \cos \theta) + E_3 D_3$$

Computation of $\sin \xi$

Similarly one has to ensure that the $\mathbf{E} \cdot \mathbf{Y}$ and $-\mathbf{y} \cdot \mathbf{Z}$ terms are equal.

Taking $\mathbf{y} \cdot \mathbf{Z} = (\mathbf{E} \times \mathbf{S}) \cdot (\mathbf{S} \times \mathbf{Y})$

since $\mathbf{A} \cdot (\mathbf{B} \times \mathbf{C}) = (\mathbf{A} \times \mathbf{B}) \cdot \mathbf{C}$

we have $\mathbf{y} \cdot \mathbf{Z} = [(\mathbf{E} \times \mathbf{S}) \times \mathbf{S}] \cdot \mathbf{Y}$

and find:

$$\mathbf{y} \cdot \mathbf{Z} = -\mathbf{E} \cdot \mathbf{Y} = -\sin \xi$$

Similarly computation of $\sin \xi$ is made as:

$$\sin \xi = \mathbf{E} \cdot \mathbf{Y}$$

thus $\sin \xi = \mathbf{E} \cdot (\mathbf{M} \times \mathbf{S})$

yet $\sin \xi = E_1(M_2S_3 - M_3S_2) + E_2(M_3S_1 - M_1S_3) + E_3(M_1S_2 - M_2S_1)$

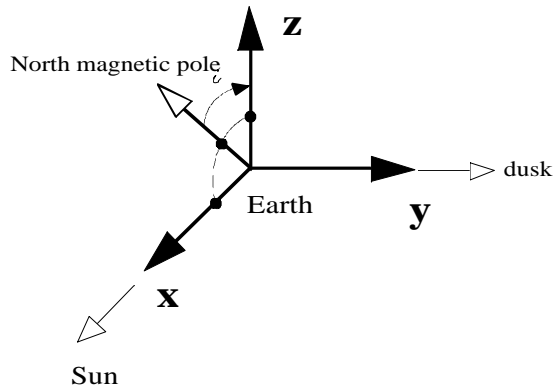
since $\mathbf{M} = ((D_1\cos\theta - D_2\sin\theta) , (D_1\sin\theta + D_2\cos\theta) , D_3)$

finally we have:

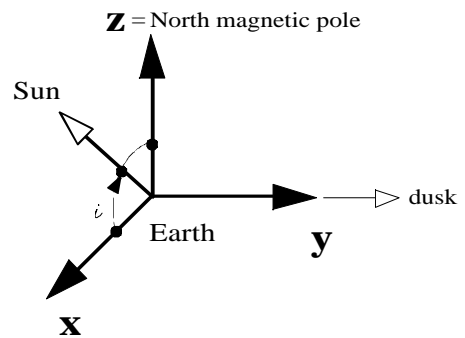
$$\begin{aligned} \sin \xi = & E_1[(D_1\sin\theta + D_2\cos\theta)S_3 - D_3S_2] \\ & + E_2[D_3S_1 - (D_1\cos\theta - D_2\sin\theta)S_3] \\ & + E_3[(D_1\cos\theta - D_2\sin\theta)S_2 - (D_1\sin\theta + D_2\cos\theta)S_1] \end{aligned}$$

10) GSM to SM transformations

Geocentric Solar Magnetospheric



Solar Magnetic



The GSM and SM system have the Y-axis in common, then the transformation matrix is a simple rotation of μ angle, which is named the dipole tilt angle.

Thus the transformation matrix of any vector V is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)} = \begin{pmatrix} \cos \mu & 0 & -\sin \mu \\ 0 & 1 & 0 \\ \sin \mu & 0 & \cos \mu \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)}$$

and the inverse transformation is obviously:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)} = \begin{pmatrix} \cos \mu & 0 & \sin \mu \\ 0 & 1 & 0 \\ -\sin \mu & 0 & \cos \mu \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

μ can be obtained from $\sin \mu = \delta \cdot \mathbf{D}$, where δ is the direction of the sun and \mathbf{D} the dipole direction, both in GEO system for instance.

δ can be computed from GEI to GEO transformation given in § III-2; then we have:

$$\delta = (\delta_1, \delta_2, \delta_3) = (S_1 \cos \theta + S_2 \sin \theta, -S_1 \sin \theta + S_2 \cos \theta, S_3)$$

where S is the direction of the Sun in GEI system, computed from CSUNDI subroutine, such as the Greenwich Mean Sidereal Time θ .

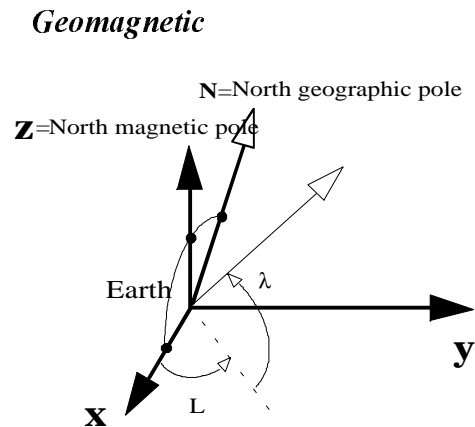
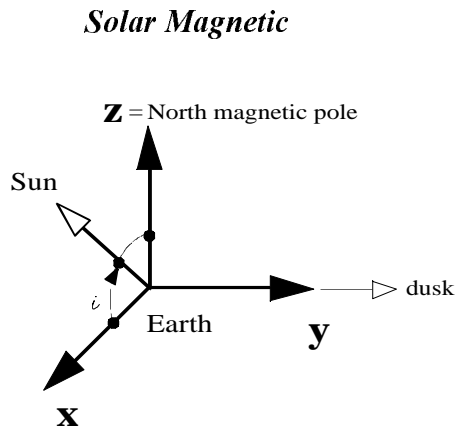
D is obtained from the International Geomagnetic Reference Field (IGRF); practically, D is computed for a given time and year from CDIPDI subroutine; value for 1965.0 is:

$$D = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

finally, rotation matrix elements are:

$$\begin{aligned} \sin \mu &= \delta_1 D_1 + \delta_2 D_2 + \delta_3 D_3 \\ \cos \mu &= (1 - \sin^2 \mu)^{1/2} \end{aligned}$$

11) SM to MAG transformations



The SM and MAG system have the Z-axis in common, then the transformation matrix is a simple rotation of φ angle, thus the transformation matrix of any vector V is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} \cos \varphi & \sin \varphi & 0 \\ -\sin \varphi & \cos \varphi & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

and the inverse transformation is obviously:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)} = \begin{pmatrix} \cos \varphi & -\sin \varphi & 0 \\ \sin \varphi & \cos \varphi & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)}$$

Nevertheless the angle φ is not derivable from a simple equation.

To compute the rotation terms of transformation matrix, we use the MAG to GEO and the GEO to SM matrix transformations, given in § III-12 and III-13.

The first transformation is noted (see § III-12):

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} X1 & X2 & X3 \\ Y1 & Y2 & Y3 \\ D1 & D2 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)}$$

with $\mathbf{X} = (\mathbf{Y} \times \mathbf{D})$ in GEO System
and $\mathbf{Y} = (\mathbf{N} \times \mathbf{D}) / |\mathbf{N} \times \mathbf{D}|$

\mathbf{N} and \mathbf{D} vector are respectively the North geographic pole and the magnetic dipole in geographic system.

Practically, geographic dipole direction \mathbf{D} is computed for a given time and year from CDIPDI subroutine; value for 1965.0 is:

$$\mathbf{D} = (D1, D2, D3) = (0.06859, -0.18602, 0.98015)$$

we deduce from $\mathbf{Y} = (\mathbf{N} \times \mathbf{D}) / |\mathbf{N} \times \mathbf{D}|$:

$$\mathbf{Y} = (-D2, D1, 0) / (D1^2 + D2^2)^{1/2}$$

The second transformation is noted (see § III-13):

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} x1 & y1 & D1 \\ x2 & y2 & D2 \\ x3 & y3 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

with $\mathbf{x} = (\mathbf{y} \times \mathbf{D})$ in GEO system
 $\mathbf{y} = (\mathbf{D} \times \mathbf{\delta}) / |\mathbf{D} \times \mathbf{\delta}|$

$\mathbf{\delta}$ is the direction of the sun in GEO system, which can be computed from GEI to GEO transformation given in § III-2, then we have:

$$\mathbf{\delta} = (\delta1, \delta2, \delta3) = (S1 \cos \theta + S2 \sin \theta, -S1 \sin \theta + S2 \cos \theta, S3)$$

S is the direction of the Sun in GEI system, computed from CSUNDI subroutine, such as the Greenwich Mean Sideral Time θ .

Knowing all elements for MAG to GEO and GEO to SM transformations, we can write the SM to MAG transformation as:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} X1 & X2 & X3 \\ Y1 & Y2 & Y3 \\ D1 & D2 & D3 \end{pmatrix} \begin{pmatrix} x1 & y1 & D1 \\ x2 & y2 & D2 \\ x3 & y3 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

which give

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} \mathbf{X} \cdot \mathbf{x} & \mathbf{X} \cdot \mathbf{y} & \mathbf{X} \cdot \mathbf{D} \\ \mathbf{Y} \cdot \mathbf{x} & \mathbf{Y} \cdot \mathbf{y} & \mathbf{Y} \cdot \mathbf{D} \\ \mathbf{D} \cdot \mathbf{x} & \mathbf{D} \cdot \mathbf{y} & \mathbf{D} \cdot \mathbf{D} \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

In other hand we can write the following equivalences:

$$\begin{aligned} \mathbf{X} \cdot \mathbf{D} &= (\mathbf{Y} \times \mathbf{D}) \cdot \mathbf{D} = \mathbf{Y} \cdot (\mathbf{D} \times \mathbf{D}) = 0 \\ \mathbf{Y} \cdot \mathbf{D} &= (\mathbf{N} \times \mathbf{D}) \cdot \mathbf{D} / |\mathbf{N} \times \mathbf{D}| = \mathbf{N} \cdot (\mathbf{D} \times \mathbf{D}) / |\mathbf{N} \times \mathbf{D}| = 0 \\ \mathbf{D} \cdot \mathbf{x} &= \mathbf{D} \cdot (\mathbf{y} \times \mathbf{D}) = -\mathbf{D} \cdot (\mathbf{D} \times \mathbf{y}) = -(\mathbf{D} \times \mathbf{D}) \cdot \mathbf{y} = 0 \\ \mathbf{D} \cdot \mathbf{y} &= \mathbf{D} \cdot (\mathbf{D} \times \mathbf{b}) / |\mathbf{D} \times \mathbf{b}| = (\mathbf{D} \times \mathbf{D}) \cdot \mathbf{b} / |\mathbf{D} \times \mathbf{b}| = 0 \\ \mathbf{D} \cdot \mathbf{D} &= 1 \end{aligned}$$

then we have the following matrix:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} \mathbf{X} \cdot \mathbf{x} & \mathbf{X} \cdot \mathbf{y} & 0 \\ \mathbf{Y} \cdot \mathbf{x} & \mathbf{Y} \cdot \mathbf{y} & 0 \\ 0 & 0 & 1 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

Computation of $\cos \varphi$

We have to equalize $\mathbf{X} \cdot \mathbf{x}$ and $\mathbf{Y} \cdot \mathbf{y}$ terms as $\cos \varphi$.

Taking
$$\mathbf{X} \cdot \mathbf{x} = (\mathbf{Y} \times \mathbf{D}) \cdot (\mathbf{y} \times \mathbf{D}) = -(\mathbf{Y} \times \mathbf{D}) \cdot (\mathbf{D} \times \mathbf{y})$$

$$\mathbf{X} \cdot \mathbf{x} = -(\mathbf{Y} \times \mathbf{D}) \cdot \mathbf{D} \cdot \mathbf{y}$$

and since \mathbf{Y} and \mathbf{D} are perpendicular:

$\mathbf{X} \cdot \mathbf{x} = \mathbf{Y} \cdot \mathbf{y} = \cos \varphi$
--

coordinates of \mathbf{Y} axe is given in § III-12 as:

$$\mathbf{Y} = (-D_2, D_1, 0) / (D_1^2 + D_2^2)^{1/2}$$

and

$$\mathbf{y} = (\mathbf{D} \times \mathbf{d}) / |\mathbf{D} \times \mathbf{d}|$$

then we have:

$$\cos \varphi = \mathbf{Y} \cdot \mathbf{y} = \mathbf{Y} \cdot (\mathbf{D} \times \mathbf{d}) / |\mathbf{D} \times \mathbf{d}| = (\mathbf{Y} \times \mathbf{D}) \cdot \mathbf{d} / |\mathbf{D} \times \mathbf{d}|$$

The $(\mathbf{Y} \times \mathbf{D})$ vector can be expanded as:

$$(\mathbf{Y} \times \mathbf{D}) = (D_1 D_3, D_2 D_3, -(D_1^2 + D_2^2)) / (D_1^2 + D_2^2)^{1/2}$$

and we deduce:

$\cos \varphi = [(D_1 D_3 d_1 + D_2 D_3 d_2 - (D_1^2 + D_2^2) d_3)] / Q$
--

with

$$Q = (D_1^2 + D_2^2)^{1/2} \cdot [(D_2 d_3 - D_3 d_2)^2 + (D_3 d_1 - D_1 d_3)^2 + (D_1 d_2 - D_2 d_1)^2]^{1/2}$$

$$(d_1, d_2, d_3) = ((S_1 \cos \theta + S_2 \sin \theta), (-S_1 \sin \theta + S_2 \cos \theta), S_3)$$

Computation of $\sin \varphi$

We have to equalize $\mathbf{X} \cdot \mathbf{y}$ and $-\mathbf{Y} \cdot \mathbf{x}$ terms as $\sin \varphi$.

Taking $\mathbf{X} \cdot \mathbf{y} = (\mathbf{Y} \times \mathbf{D}) \cdot \mathbf{y}$

and $-\mathbf{Y} \cdot \mathbf{x} = -\mathbf{Y} \cdot (\mathbf{y} \times \mathbf{D}) = \mathbf{Y} \cdot (\mathbf{D} \times \mathbf{y}) = (\mathbf{Y} \times \mathbf{D}) \cdot \mathbf{y}$

yet we have well:

$$\mathbf{X} \cdot \mathbf{y} = -\mathbf{Y} \cdot \mathbf{x} = \sin \varphi$$

$\sin \varphi$ is then computed from $\sin \varphi = (\mathbf{Y} \times \mathbf{D}) \cdot \mathbf{y}$

from above we have:

$$(\mathbf{Y} \times \mathbf{D}) = (D_1 D_3, D_2 D_3, -(D_1^2 + D_2^2)) / (D_1^2 + D_2^2)^{1/2}$$

and

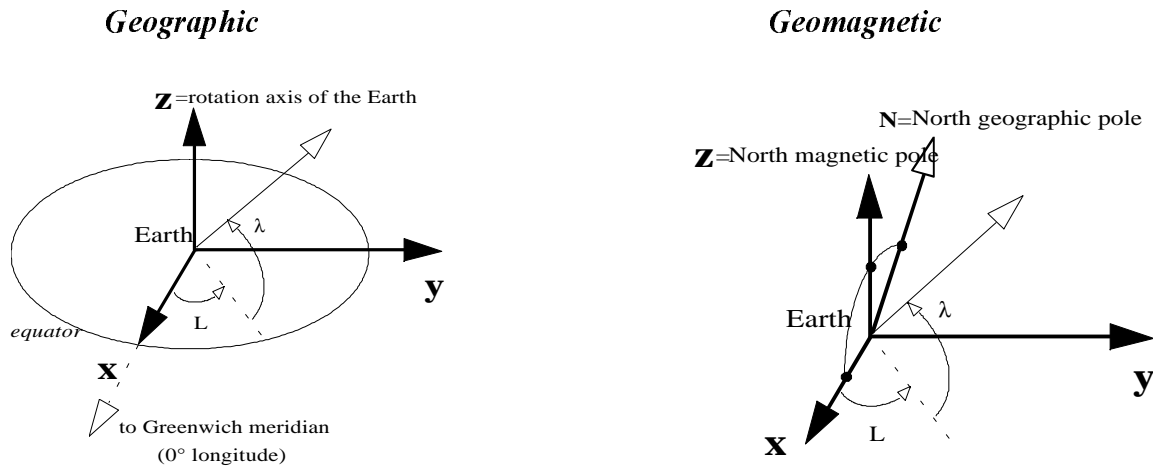
$$\mathbf{y} = (\mathbf{D} \times \mathbf{d}) / |\mathbf{D} \times \mathbf{d}|$$

we deduce:

$$\sin \varphi = (D_2 d_1 - D_1 d_2) / Q$$

with Q such as above.

12) GEO to MAG transformations



In geomagnetic coordinates, Z-axis is parallel to the magnetic dipole axis \mathbf{D} , and the Y-axis is perpendicular to the North geographic pole \mathbf{N} , then we have in geographic system:

$$\begin{aligned} \mathbf{Z} &= \mathbf{D} \\ \mathbf{Y} &= (\mathbf{N} \times \mathbf{D}) / |\mathbf{N} \times \mathbf{D}| \\ \text{and } \mathbf{X} &= \mathbf{Y} \times \mathbf{D} \end{aligned}$$

The geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude.

Practically, geographic dipole direction \mathbf{D} is computed for a given time and year from CDIPDI subroutine; value for 1965.0 is:

$$\mathbf{D} = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

we deduce:

$$\mathbf{Y} = (-D_2, D_1, 0) / (D_1^2 + D_2^2)^{1/2}$$

and:

$$\mathbf{X} = (Y_2 D_3, -Y_1 D_3, Y_1 D_2 - Y_2 D_1)$$

Thus the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} X_1 & X_2 & X_3 \\ Y_1 & Y_2 & Y_3 \\ D_1 & D_2 & D_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GEO)}$$

Similarly the transformation from system MAG to GEO is:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} X1 & Y1 & D1 \\ X2 & Y2 & D2 \\ X3 & Y3 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)}$$

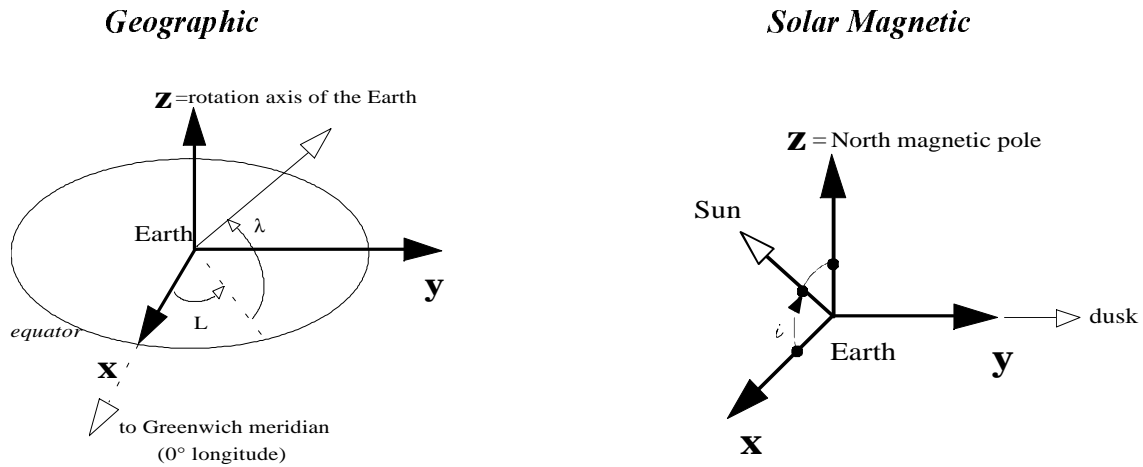
For IGRF epoch 1965.0, we have for instance:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)} = \begin{pmatrix} 0.33908 & -0.91963 & -0.19826 \\ 0.93825 & 0.34595 & 0. \\ 0.06859 & -0.18602 & 0.98015 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)}$$

and similarly:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} 0.33908 & 0.93825 & 0.06859 \\ -0.91963 & 0.34595 & -0.18602 \\ -0.19826 & 0. & 0.98015 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(MAG)}$$

13) GEO to SM transformations



In GEO system, the direction of the Z-axis of SM system is the dipole direction \mathbf{D} . The geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude, thus:

$$\mathbf{D} = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

Practically, \mathbf{D} is computed for a given time and year from CDIPDI subroutine.

We can then deduce in GEO system the Y-axis of SM system as:

$$\mathbf{y} = (\mathbf{D} \times \mathbf{s}) / |\mathbf{D} \times \mathbf{s}|$$

where \mathbf{s} is the direction of the sun in GEO system, which can be computed from GEI to GEO transformation given in § III-2, then we have:

$$\mathbf{s} = (s_1, s_2, s_3) = ((S_1 \cos \theta + S_2 \sin \theta), (-S_1 \sin \theta + S_2 \cos \theta), S_3)$$

\mathbf{S} is the direction of the Sun in GEI system, computed from CSUNDI subroutine, such as the Greenwich Mean Sidereal Time θ ; normalizing factors occurs because \mathbf{D} and \mathbf{s} are not necessarily perpendicular.

The third axis \mathbf{X} is computed from:

$$\mathbf{x} = (\mathbf{y} \times \mathbf{D})$$

and the GEO to SM transformation is given by:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)} = \begin{pmatrix} x1 & x2 & x3 \\ y1 & y2 & y3 \\ D1 & D2 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)}$$

Similarly the SM to GEO transformation is given by:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} x1 & y1 & D1 \\ x2 & y2 & D2 \\ x3 & y3 & D3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(SM)}$$

with respectively:

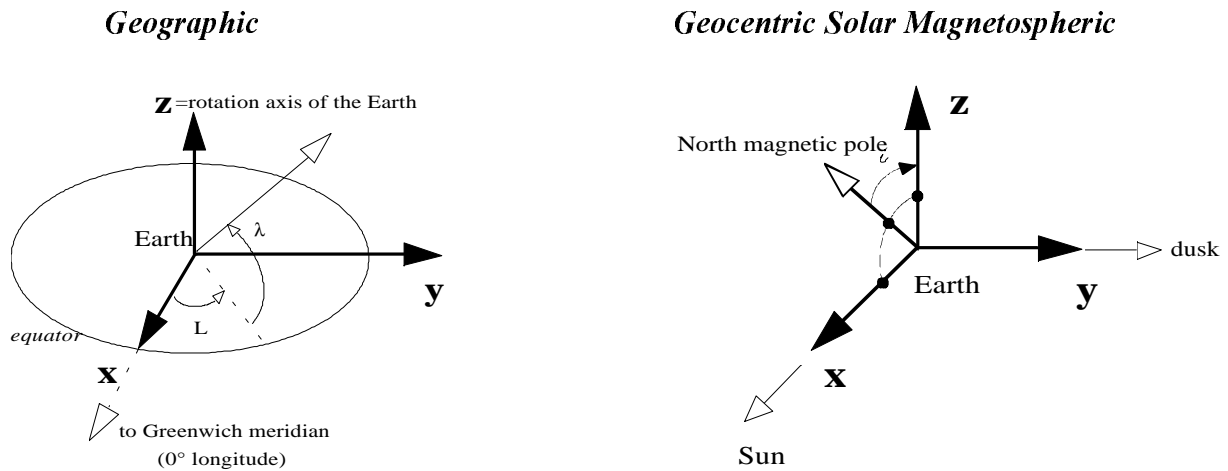
$$\begin{pmatrix} x1 \\ x2 \\ x3 \end{pmatrix} = \begin{pmatrix} y2D3 & - & y3D2 \\ y3D1 & - & y1D3 \\ y1D2 & - & y2D1 \end{pmatrix}$$

$$\begin{pmatrix} y1 \\ y2 \\ y3 \end{pmatrix} = \begin{pmatrix} D2\delta3 & - & D3\delta2 \\ D3\delta1 & - & D1\delta3 \\ D1\delta2 & - & D2\delta1 \end{pmatrix} \cdot 1 / Q$$

$$Q = [(D2\delta3 - D3\delta2)^2 + (D3\delta1 - D1\delta3)^2 + (D1\delta2 - D2\delta1)^2]^{1/2}$$

$$\delta = (\delta1, \delta2, \delta3) = ((S1\cos\theta + S2\sin\theta) , (-S1\sin\theta + S2\cos\theta) , S3)$$

14) GEO to GSM transformations



In GEO system, the direction of the X-axis of GSM system is the direction of the sun δ , which can be computed from GEI to GEO transformation given in § III-2; then we have:

$$\mathbf{x} = \delta = (\delta_1, \delta_2, \delta_3) = ((S_1 \cos \theta + S_2 \sin \theta), (-S_1 \sin \theta + S_2 \cos \theta), S_3)$$

S is the direction of the Sun in GEI system, computed from CSUNDI subroutine, such as the Greenwich Mean Sidereal Time θ .

The Y axis in GEO system can be deduced from:

$$\mathbf{y} = (\mathbf{D} \times \delta) / |\mathbf{D} \times \delta|$$

where \mathbf{D} is the dipole direction; the geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude, thus:

$$\mathbf{D} = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

Practically, \mathbf{D} is computed for a given time and year from CDIPDI subroutine; normalizing factors in cross product occurs because \mathbf{D} and δ are not necessarily perpendicular.

And finally the third axis Z is computed from:

$$\mathbf{z} = \delta \times \mathbf{y}$$

and the GEO to GSM transformation is given by:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)} = \begin{pmatrix} x1 & x2 & x3 \\ y1 & y2 & y3 \\ z1 & z2 & z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)}$$

Similarly the GSM to GEO transformation is given by:

$$\begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} x1 & y1 & z1 \\ x2 & y2 & z2 \\ x3 & y3 & z3 \end{pmatrix} \begin{pmatrix} V1 \\ V2 \\ V3 \end{pmatrix}_{(GSM)}$$

with respectively:

$$\begin{pmatrix} x1 \\ x2 \\ x3 \end{pmatrix} = \begin{pmatrix} \delta1 \\ \delta2 \\ \delta3 \end{pmatrix}$$

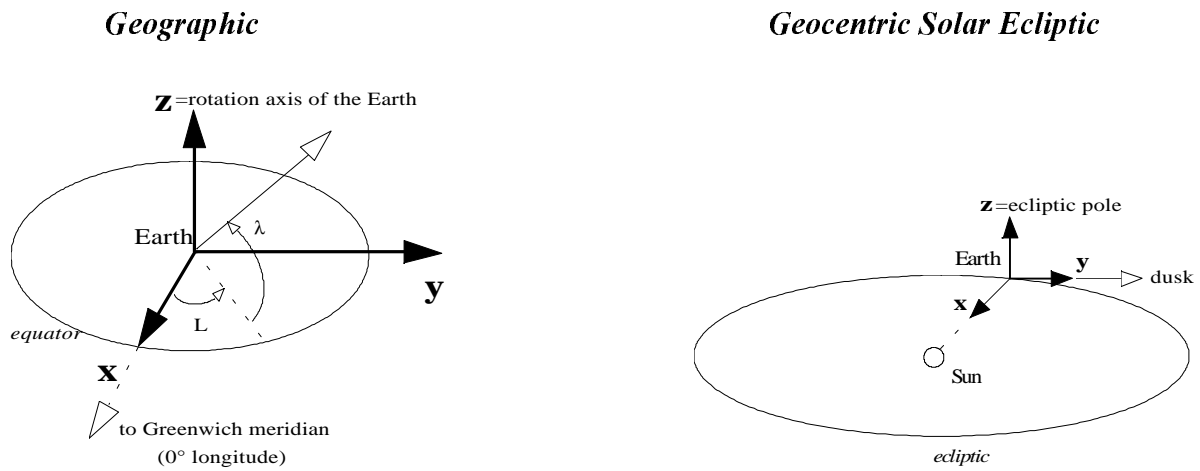
$$\begin{pmatrix} y1 \\ y2 \\ y3 \end{pmatrix} = \begin{pmatrix} D2\delta3 & - & D3\delta2 \\ D3\delta1 & - & D1\delta3 \\ D1\delta2 & - & D2\delta1 \end{pmatrix} \cdot 1 / Q$$

$$\begin{pmatrix} z1 \\ z2 \\ z3 \end{pmatrix} = \begin{pmatrix} x2y3 & - & x3y2 \\ x3y1 & - & x1y3 \\ x1y2 & - & x2y1 \end{pmatrix}$$

$$Q = [(D2\delta3 - D3\delta2)^2 + (D3\delta1 - D1\delta3)^2 + (D1\delta2 - D2\delta1)^2]^{1/2}$$

$$\delta = (\delta1, \delta2, \delta3) = ((S1\cos\theta + S2\sin\theta) , (-S1\sin\theta + S2\cos\theta) , S3)$$

15) GEO to GSE transformations



In GEO system, the direction of the X-axis of GSE system is the direction of the sun δ , which can be computed from GEI to GEO transformation given in § III-2; then we have:

$$\mathbf{X}=\delta=(\delta_1, \delta_2, \delta_3)=((S_1\cos\theta + S_2\sin\theta), (-S_1\sin\theta + S_2\cos\theta), S_3)$$

S is the direction of the Sun in GEI system, computed from CSUNDI subroutine, such as the Greenwich Mean Sideral Time θ .

The Z axis is the direction of the ecliptic pole, which is a known constant value in GEI system:

$$\mathbf{E}=(E_1, E_2, E_3)=(0, -0.398, 0.917)$$

from GEI to GEO transformation we have:

$$\mathbf{Z}=\mathcal{E}=(\mathcal{E}_1, \mathcal{E}_2, \mathcal{E}_3)=((E_1\cos\theta + E_2\sin\theta), (-E_1\sin\theta + E_2\cos\theta), E_3)$$

And finally the Y axis in GEO system can be deduced from:

$$\mathbf{Y}=\mathcal{E} \times \delta$$

and the GEO to GSE transformation is given by:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GSE)} = \begin{pmatrix} \delta_1 & \delta_2 & \delta_3 \\ Y_1 & Y_2 & Y_3 \\ \varepsilon_1 & \varepsilon_2 & \varepsilon_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GEO)}$$

Similarly the GSE to GEO transformation is given by:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} \delta_1 & Y_1 & \varepsilon_1 \\ \delta_2 & Y_2 & \varepsilon_2 \\ \delta_3 & Y_3 & \varepsilon_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GSE)}$$

with respectively:

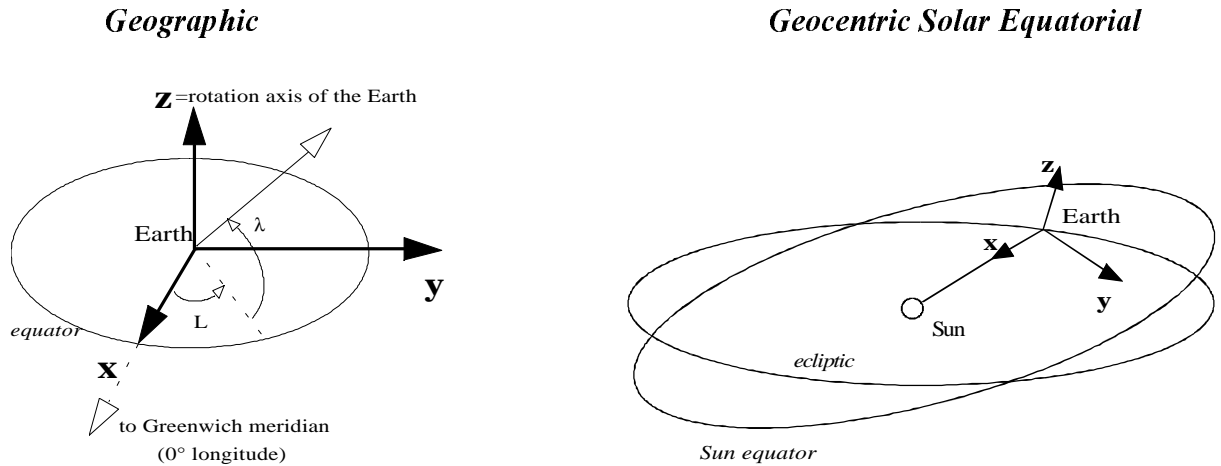
$$\begin{pmatrix} Y_1 \\ Y_2 \\ Y_3 \end{pmatrix} = \begin{pmatrix} \varepsilon_2 \delta_3 & - \varepsilon_3 \delta_2 \\ \varepsilon_3 \delta_1 & - \varepsilon_1 \delta_3 \\ \varepsilon_1 \delta_2 & - \varepsilon_2 \delta_1 \end{pmatrix}$$

$$\delta = (\delta_1, \delta_2, \delta_3) = ((S_1 \cos \theta + S_2 \sin \theta), (-S_1 \sin \theta + S_2 \cos \theta), S_3)$$

$$\varepsilon = (\varepsilon_1, \varepsilon_2, \varepsilon_3) = ((E_1 \cos \theta + E_2 \sin \theta), (-E_1 \sin \theta + E_2 \cos \theta), E_3)$$

$$\mathbf{E} = (E_1, E_2, E_3) = (0, -0.398, 0.917)$$

16) GEO to GSEQ transformations



In GEO system, the direction of the X-axis of GSEQ system is the direction of the sun δ , which can be computed from GEI to GEO transformation given in § III-2; then we have:

$$\mathbf{x} = \delta = (\delta_1, \delta_2, \delta_3) = (S_1 \cos \theta + S_2 \sin \theta, -S_1 \sin \theta + S_2 \cos \theta, S_3)$$

S is the direction of the Sun in GEI system, computed from CSUNDI subroutine, such as the Greenwich Mean Sidereal Time θ .

The Sun equator plane is defined from the direction of the rotation axis of the SUN which is in GEI system a known constant value:

$$\mathbf{R} = (R_1, R_2, R_3) = (0.122, -0.424, 0.899)$$

from GEI to GEO transformation we obtain this vector in GEO system as:

$$\mathbf{R} = (\mathcal{R}_1, \mathcal{R}_2, \mathcal{R}_3) = (R_1 \cos \theta + R_2 \sin \theta, -R_1 \sin \theta + R_2 \cos \theta, R_3)$$

Then the Y axis in GEO system can be deduced from:

$$\mathbf{y} = \mathbf{R} \times \delta / |\mathbf{R} \times \delta|$$

normalizing factors in cross product occurs because \mathcal{R} and δ are not necessarily perpendicular.

The third axis Z is computed from:

$$\mathbf{z} = \delta \times \mathbf{y}$$

and the GEO to GSEQ transformation is given by:

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSEQ)} = \begin{pmatrix} x 1 & x 2 & x 3 \\ y 1 & y 2 & y 3 \\ z 1 & z 2 & z 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GEO)}$$

Similarly the GEQ to GEO transformation is given by:

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} x 1 & y 1 & z 1 \\ x 2 & y 2 & z 2 \\ x 3 & y 3 & z 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GSEQ)}$$

with respectively:

$$\begin{pmatrix} x 1 \\ x 2 \\ x 3 \end{pmatrix} = \begin{pmatrix} \delta 1 \\ \delta 2 \\ \delta 3 \end{pmatrix}$$

$$\begin{pmatrix} y 1 \\ y 2 \\ y 3 \end{pmatrix} = \begin{pmatrix} \rho 2 \delta 3 - \rho 3 \delta 2 \\ \rho 3 \delta 1 - \rho 1 \delta 3 \\ \rho 1 \delta 2 - \rho 2 \delta 1 \end{pmatrix} \cdot 1 / Q$$

$$\begin{pmatrix} z 1 \\ z 2 \\ z 3 \end{pmatrix} = \begin{pmatrix} \delta 2 y 3 - \delta 3 y 2 \\ \delta 3 y 1 - \delta 1 y 3 \\ \delta 1 y 2 - \delta 2 y 1 \end{pmatrix}$$

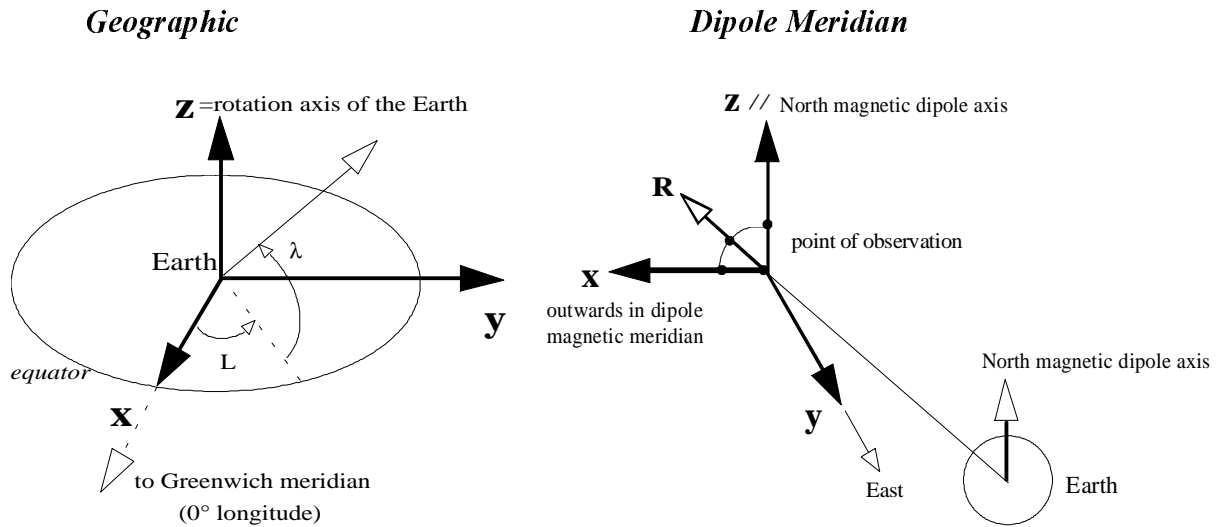
$$Q = [(\rho 2 \delta 3 - \rho 3 \delta 2)^2 + (\rho 3 \delta 1 - \rho 1 \delta 3)^2 + (\rho 1 \delta 2 - \rho 2 \delta 1)^2]^{1/2}$$

$$\delta = (\delta 1, \delta 2, \delta 3) = ((S_1 \cos \theta + S_2 \sin \theta), (-S_1 \sin \theta + S_2 \cos \theta), S_3)$$

$$\rho = (\rho 1, \rho 2, \rho 3) = ((R_1 \cos \theta + R_2 \sin \theta), (-R_1 \sin \theta + R_2 \cos \theta), R_3)$$

$$\mathbf{R} = (R_1, R_2, R_3) = (0.122, -0.424, 0.899)$$

17) GEO to DM transformations



Dipole meridian system is a local coordinate system, and varies with the position of the point of observation relative to the centre of the Earth; this position is noted in GEO system as:

$$\mathbf{R} = (R_1, R_2, R_3)$$

To transform GEO coordinate to DM coordinates, we need the dipole position in GEO system which is the Z axis of DM system. The geographic coordinates of the dipole axis can be known, for instance for IGRF epoch 1965, as 11.435° colatitude and -69.761° east longitude, thus:

$$\mathbf{Z} = \mathbf{D} = (D_1, D_2, D_3) = (0.06859, -0.18602, 0.98015)$$

Practically, \mathbf{D} is computed for a given time and year from CDIPDI subroutine.

We can deduce then the Y-axis of DM system in GEO coordinates as:

$$\mathbf{Y} = \mathbf{D} \times \mathbf{R} / |\mathbf{D} \times \mathbf{R}|$$

normalizing factors occurs because \mathbf{D} and \mathbf{R} are not necessarily perpendicular and because \mathbf{R} is not a unit vector.

The third axis X is deduced from:

$$\mathbf{X} = \mathbf{Y} \times \mathbf{D}$$

All coordinates of X-Y-Z axis of DM system in GEO coordinates being known, the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(DM)} = \begin{pmatrix} X_1 & X_2 & X_3 \\ Y_1 & Y_2 & Y_3 \\ D_1 & D_2 & D_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GEO)}$$

Similarly the transformation from system DM to GEO is:

$$\begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} X_1 & Y_1 & D_1 \\ X_2 & Y_2 & D_2 \\ X_3 & Y_3 & D_3 \end{pmatrix} \begin{pmatrix} V_1 \\ V_2 \\ V_3 \end{pmatrix}_{(DM)}$$

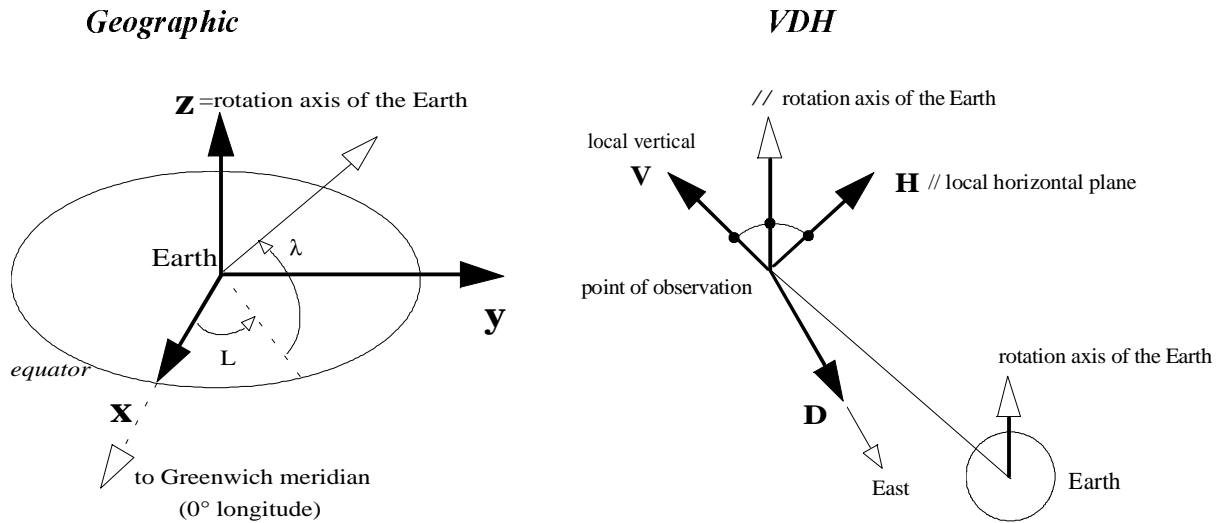
with respectively:

$$\begin{pmatrix} X_1 \\ X_2 \\ X_3 \end{pmatrix} = \begin{pmatrix} Y_2 D_3 & - & Y_3 D_2 \\ Y_3 D_1 & - & Y_1 D_3 \\ Y_1 D_2 & - & Y_2 D_1 \end{pmatrix}$$

$$\begin{pmatrix} Y_1 \\ Y_2 \\ Y_3 \end{pmatrix} = \begin{pmatrix} D_2 R_3 & - & D_3 R_2 \\ D_3 R_1 & - & D_1 R_3 \\ D_1 R_2 & - & D_2 R_1 \end{pmatrix} \cdot 1 / Q$$

$$Q = \left[(D_2 R_3 - D_3 R_2)^2 + (D_3 R_1 - D_1 R_3)^2 + (D_1 R_2 - D_2 R_1)^2 \right]^{1/2}$$

18) GEO to VDH transformations



VDH system is a local coordinate system, and varies with the position of the point of observation relative to the centre of the Earth; this positions is noted in GEO system as:

$$\mathbf{R} = (R_1, R_2, R_3)$$

and we have directly

$$\mathbf{V} = \mathbf{R} / (R_1^2 + R_2^2 + R_3^2)^{1/2}$$

Since D is perpendicular to V and to the rotation axis of the Earth, we have:

$$\mathbf{D} = \mathbf{Z}_E \times \mathbf{R} / |\mathbf{Z}_E \times \mathbf{R}|$$

which give:

$$\mathbf{D} = \begin{pmatrix} -R_2 \\ R_1 \\ 0 \end{pmatrix} \cdot 1 / (R_1^2 + R_2^2)^{1/2}$$

The third axis H is deduced from:

$$\mathbf{H} = \mathbf{V} \times \mathbf{D}$$

which give:

$$\mathbf{H} = \begin{pmatrix} -R_1 R_3 \\ -R_2 R_3 \\ R_1^2 + R_2^2 \end{pmatrix} \cdot 1 / Q$$

with

$$Q = [(R_1^2 + R_2^2)(R_1^2 + R_2^2 + R_3^2)]^{1/2}$$

All coordinates of X-Y-Z axis of VDH system in GEO coordinates being known, the transform matrix of any vector \mathbf{V} is:

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(VDH)} = \begin{pmatrix} V 1 & V 2 & V 3 \\ D 1 & D 2 & D 3 \\ H 1 & H 2 & H 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GEO)}$$

Similarly the transformation from system VDH to GEO is:

$$\begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(GEO)} = \begin{pmatrix} V 1 & D 1 & H 1 \\ V 2 & D 2 & H 2 \\ V 3 & D 3 & H 3 \end{pmatrix} \begin{pmatrix} V 1 \\ V 2 \\ V 3 \end{pmatrix}_{(VDH)}$$

IV- CALENDAR CONVERSIONS

1) general remarks

Most space missions deliver or use for their processing data date and time in various format, such as decimal Julian day, month-day in the month, year, decimal hours, hour-min-sec, and so on. In order to easily manipulate all these quantities, and to get transformations between else, special calendar conversions, described below, has been added to ROCOT library.

2) list of calendar conversions

name *work*

cleapy compute if a given year is or not a leap year.

codoty compute day of the year from given month, day in the month and year.

cdfdot compute date as month, day in the month, from given day of the year and year

cjud50 compute Julian day from 1/1/1950 from given month, day in the month and year.

cdfjud compute date as month, day in the month and year from given 1950 Julian day.

codech compute decimal hour from given hour, minute and second.

cotime compute hour, minute and second from given decimal hour.

3) definition of conversions

compute if a given year is or not a leap year (cleapy)

A given year (as 1990) is a leap year if this number is divisible by 4; if the year is a secular year (as 1900 or 2000), this year is a leap year if this number is divisible by 400.

For instance, 1988, 1992, 2000 are leap years, 1900, 1989, 2100 are not.

compute day of the year from given month, day in the month and year (codoty)

this conversion compute the day of the year, equal to 1 for January 1, for a date given as month, day in the month, and year.

For instance, the day of the year of October 17, 1990 is 290. This conversion counts the number of the day in a month, taking into account leap year since February has not a constant number of days (using cleapy).

compute date as month, day in the month, from given day of the year and year (cdfdot)

This is the inverse conversion of codoty, since one compute the corresponding month and day in the month for a given day of the year, and a given year. Year is necessary to know if the year is or not a leap year (using cleapy).

compute Julian day from given month, day in the month and year (cjud50)

From a given date as month, day, year, this conversion compute the Julian day , i.e. day number from January 1, 1950, with Julian day equal to 1 for January 1, 1950 (use codoty and cleapy conversions).

compute date as month, day in the month and year from given 1950 Julian day (cdfjud)

This is the inverse conversion of cjud50, since one compute for a given Julian day the corresponding date as month, day in the month and year (use cleapy and cdfdot conversions).

compute decimal hour from given hour, minute and second (codech)

This conversion makes the operation $\text{hour} + \text{minute}/60. + \text{second}/3600.$

compute hour, minute and second from given decimal hour (cotime)

Inverse of codech.

V- ROCOT LIBRARY

1) general remarks

ROCOT library delivers 4 kinds of subroutines:

a) "Read and check" subroutines, such as `rxxxxxx` (`redate`, `retime`), which query and read on the terminal date and time, with automatic check of good format (no hour greater than 23, or minutes greater than 59, and so on), useful to write an interactive user program where date and time are an input.

b) "Computation" subroutines, such as `xxxxxxx` (`codoty`, `cotime` etc..). that only do computations without any particular previous call. These are mainly concerned with calendar conversions and are useful for dates and times given in various format.

One has include in this category the particular '`ctimpa`' subroutine which set for a given date and time all the time varying quantities needed for the transformations matrix (results are stored in 15 common transparent to the user).

c) "Give" subroutines, such as `gxxxxxx` (`gsundi`, `gdipdi`, etc..) which give in the output arguments useful parameters as direction of the Sun, direction of the dipole etc... in GEI and GEO systems; subroutine `ctimpa` must be called before using theses subroutines at each times where the date and time are changed;

d) "Transformations" subroutines, such as `txxxxxxx` (`tgeigeo`, `tgeogei`, `tmaggs`, `tgsmmag`, etc..) which transforms input cartesian coordinates system into an another one following mathematical expressions given in section III; as above, subroutine `ctimpa` must be called before using theses subroutines at each times where the date and time are changed. Note that the name have 7 characters rather 6, but it allows to avoid confusion with any equivalent subroutine in Tsyganenko magnetic field model which could be used simultaneously with Rocotlib; most of Fortran compiler accept it.

Section VI give an example of Fortran user program, directions for use, example of installation of the library on a UNIX system, and provide a test program code source to check the validity of the library.

2) description of "Read and check" subroutines

redate

subroutine redate(imonth,iday,iyear)

read_date from terminal input and check validity

test if imonth is not greater than 12, test if iday is not greater then length of the month, taking into account the leap years; year must be greater or equal to 1900; use cleapy subroutine.

input : imonth,iday,iyear

output: print error if date is not valid, and ask again

retime

subroutine retime(ih,im,is)

read_time from terminal input and check validity

ih must be between 1 and 23, im and is between 1 and 59

input : ih,im,is

output: print error if time is not valid, and ask again

3) description of "Computation" subroutines

ctimpa

subroutine `ctimpa(iyear, idoty, ihour, imin, isec)`

compute_time_parameters: prepare matrix for coordinate transformations

prepare all time varying quantities for computations of coordinate transforms of the library, and store results in 15 common statements; use `csundi` and `cdipdi` subroutines. Quantities stored in commons are below:

*sin and cos of GMST
ecliptic pole in GEI system
direction of the rotation axis of the sun in GEI system
dipole direction in GEI system
direction of the sun in GEO system
direction of the ecliptic in GEO system
direction of the rotation axis of the sun in GEO system
cross product $M \times S$ in GEI system
cross product $E \times S$ in GEI system
cross product $R \times S$ in GEI system
cross product $R \times E$ in GEI system
cross product $D \times S$ in GEO system
cross product $E \times S$ in GEO system
cross product $R \times S$ in GEO system
computation of gei to mag vectors
computation of gei to sm vectors
computation of gei to gsm vectors
computation of gei to gseq vectors
computation of tetq angle
computation of mu angle
computation of dzeta angle
computation of phi angle
computation of geo to mag vectors
computation of geo to sm vectors
computation of geo to gsm vectors
computation of geo to gseq vectors*

input : `iyear` : year (1901-2099)
`idoty` : day of the year (1 for January 1)
`ihour, imin, isec` : hours, minutes, seconds U.T.

output: in common statements

cdipdi

subroutine cdipdi(iyear, idoty, d1, d2, d3)

compute_dipole_direction in GEO system

compute geodipole axis direction from International Geomagnetic Reference Field (IGRF) models for time interval 1965 to 1990. For time out of interval, computation is made for nearest boundary.

input : iyear : year (1965 - 1990)
output: d1, d2, d3 cartesian dipole components in GEO

csundi

subroutine csundi(iyear, idoty, ihour, imin, isec, gst, slong, sra, sdec)

compute_sun_direction in GEI system

*(from C.T. Russel, cosmic electrodynamics, v.2, 184-196, 1971)
calculates four quantities in gei system necessary for coordinate transformations dependent on sun position (and, hence, on universal time and season)*

input : iyear : year (1901-2099)
 idoty : day of the year (1 for January 1)
 ihour, imin, isec : hours, minutes, seconds U.T.

output: gst Greenwich mean sidereal time (radians)
 slong longitude along ecliptic (radians)
 sra right ascension (radians)
 sdec declination of the sun (radians)

cleapy

subroutine cleapy(iyear, ileap)

compute_leap_year with ileap=1 for leap year, 0 if not

input : iyear (ex: 1980)
output: ileap (1 or 0 if iyear is or not a leap year)

codoty

subroutine codoty(imonth, iday, iyear, idoty)

*compute_day_of_the_year with idoty=1 for January 1
(taking into account leap year); use cleapy.*

input : imonth, iday, iyear ex: 10, 17, 1990
output: idoty ex: 290

cjud50

subroutine cjud50(imonth,iday,iyear,jud50)

compute_Julian_day with jud50=1 for January 1, 1950
(taking account leap years); use codoty and cleapy

input : imonth,iday,iyear ex: 10,17,1990
output: jud50

cdfdot

subroutine cdfdot(iyear,idoty,imonth,iday)

compute_date_from_day_of_the_year and for a given year; use cleapy.

input : iyear,idoty (idoty=1 for January 1)
output: imonth,iday

cdfjud

subroutine cdfjud(jud50,imonth,iday,iyear)

compute_date_from_Julian_day with jud50=1 for January 1, 1950
use cleapy and cdfdot

input : jud50 Julian day (1= 1/1/1950)
output: imonth,iday,iyear

codech

subroutine codech(ih,im,is,dech)

compute_decimal_hour from hours, minutes, seconds

input : ih,im,is
output: dech decimal hour

cotime

subroutine cotime(dech,ih,im,is)

compute_time from decimal hour

input : dech decimal hour
output: ih,im,is

canara

```
subroutine canara(ux,uy,uz,vx,vy,vz,angle,ratio)
```

compute_angle_and_ratio between *U* and *V* vectors

```
input : ux,uy,uz  
       vx,vy,vz
```

```
output: angle=angle between U and V (radians)  
       ratio= mod(U)/mod(V)
```

4) description of "Give" subroutines

gsundi

```
subroutine gsundi(sxgei,sygei,szgei,sxgeo,sygeo,szgeo)
```

give_sun_direction in *GEI* and *GEO* system

```
input : none  
output: sxgei,sygei,szgei cartesian sun GEI coord.  
       sxgeo,sygeo,szgeo cartesian sun GEO coord.
```

gdipdi

```
subroutine gdipdi(dxgei,dygei,dzgei,dxgeo,dygeo,dzgeo)
```

give_dipole_direction in *GEI* and *GEO* system

```
input : none  
output: dxgei,dygei,dzgei cartesian dipole GEI coord.  
       dxgeo,dygeo,dzgeo cartesian dipole GEO coord.
```

gecldi

```
subroutine gecldi(exgei,eygei,ezgei,exgeo,eygeo,ezgeo)
```

give_ecliptic_direction in *GEI* and *GEO* system

```
input : none  
output: exgei,eygei,ezgei cartesian eclip. GEI coord.  
       exgeo,eygeo,ezgeo cartesian eclip. GEO coord.
```

gsrodi

subroutine gsrodi(rxgei,rygei,rzgei,rxgeo,rygeo,rzgeo)

give_sun_rotation_direction in GEI and GEO system

input : none

output: rxgei,rygei,rzgei cartesian sun rot. GEI coord.
rxgeo,rygeo,rzgeo cartesian sun rot. GEO coord.

gsunpa

subroutine gsunpa(gmst,slon,sras,sdec)

give_sun_parameter dependant of time in GEI system

input : none

output: gmst Greenwich mean sidereal time(radians)
slon longitude along ecliptic (radians)
sras right ascension (radians)
sdec declination of the sun (radians)

gdipta

subroutine gdipta(dipta)

give_dipole_tilt_angle in GSM system (radians)

input : none

output: dipta=(D,Z) angle in GSM (radians)
(dipta>0 when the north magnetic pole
is tilted toward the Sun)

gvernu

subroutine gvernu(vernu)

give_version_number of the library

input : none

output: vernu (ex 1.0)

5) description of "Transformations" subroutines

tgeigeo

subroutine tgeigeo(xgei,ygei,zgei,xgeo,ygeo,zgeo)

transforms_gei_to_geo: GEI -> GEO system

input : xgei,ygei,zgei cartesian gei coordinates
output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeogei

subroutine tgeogei(xgeo,ygeo,zgeo,xgei,ygei,zgei)

transforms_geo_to_gei: GEO -> GEI system

input : xgeo,ygeo,zgeo cartesian geo coordinates
output: xgei,ygei,zgei cartesian gei coordinates

tgeimag

subroutine tgeimag(xgei,ygei,zgei,xmag,ymag,zmag)

transforms_gei_to_mag: GEI -> MAG system

input : xgei,ygei,zgei cartesian gei coordinates
output: xmag,ymag,zmag cartesian mag coordinates

tmaggei

subroutine tmaggei(xmag,ymag,zmag,xgei,ygei,zgei)

transforms_mag_to_gei: MAG -> GEI system

input : xmag,ymag,zmag cartesian mag coordinates
output: xgei,ygei,zgei cartesian gei coordinates

tgeisma

subroutine tgeisma(xgei,ygei,zgei,xsma,ysma,zsma)

transforms_gei_to_sma: GEI -> SM system

input : xgei,ygei,zgei cartesian gei coordinates
output: xsma,ysma,zsma cartesian sma coordinates

tsmagei

subroutine tsmagei(xsma,ysma,zsma,xgei,ygei,zgei)

transforms_sma_to_gei: SM -> GEI system

input : xsma,ysma,zsma cartesian sma coordinates
output: xgei,ygei,zgei cartesian gei coordinates

tgeism

subroutine tgeigsm(xgei,ygei,zgei,xgsm,ygsm,zgsm)

transforms_gei_to_gsm: GEI -> GSM system

input : xgei,ygei,zgei cartesian gei coordinates
output: xgsm,ygsm,zgsm cartesian gsm coordinates

tgsmgei

subroutine tgsmgei(xgsm,ygsm,zgsm,xgei,ygei,zgei)

transforms_gsm_to_gei: GSM -> GEI system

input : xgsm,ygsm,zgsm cartesian gsm coordinates
output: xgei,ygei,zgei cartesian gei coordinates

tgeigse

subroutine tgeigse(xgei,ygei,zgei,xgse,ygse,zgse)

transforms_gei_to_gse: GEI -> GSE system

input : xgei,ygei,zgei cartesian gei coordinates
output: xgse,ygse,zgse cartesian gse coordinates

tgsegei

subroutine tgsegei(xgse,ygse,zgse,xgei,ygei,zgei)

transforms_gse_to_gei: GSE -> GEI system

input : xgse,ygse,zgse cartesian gse coordinates
output: xgei,ygei,zgei cartesian gei coordinates

tgeigsq

subroutine tgeigsq(xgei,ygei,zgei,xgsq,ygsq,zgsq)

transforms_gei_to_gsq: GEI -> GSEQ system

input : xgei,ygei,zgei cartesian gei coordinates
output: xgsq,ygsq,zgsq cartesian gsq coordinates

tgsqgei

subroutine tgsqgei(xgsq,ygsq,zgsq,xgei,ygei,zgei)

transforms_gsq_to_gei: GSEQ-> GEI system

input : xgsq,ygsq,zgsq cartesian gsq coordinates
output: xgei,ygei,zgei cartesian gei coordinates

tgsegsq

subroutine tgsegsq(xgse,ygse,zgse,xgsq,ygsq,zgsq)

transforms_gse_to_gsq: GSE -> GSEQ system

input : xgse,ygse,zgse cartesian gse coordinates
output: xgsq,ygsq,zgsq cartesian gsq coordinates

tgsqgse

subroutine tgsqgse(xgsq,ygsq,zgsq,xgse,ygse,zgse)

transforms_gsq_to_gse: GSEQ-> GSE system

input : xgsq,ygsq,zgsq cartesian gsq coordinates
output: xgse,ygse,zgse cartesian gse coordinates

tgsegsm

subroutine tgsegsm(xgse,ygse,zgse,xgsm,ygsm,zgsm)

transforms_gse_to_gsm: GSE -> GSM system

input : xgse,ygse,zgse cartesian gse coordinates
output: xgsm,ygsm,zgsm cartesian gsm coordinates

tgsmgse

subroutine tgsmgse(xgsm,ygsm,zgsm,xgse,ygse,zgse)

transforms_gsm_to_gse: GSM -> GSE system

input : xgsm,ygsm,zgsm cartesian gsm coordinates
output: xgse,ygse,zgse cartesian gse coordinates

tgmsma

subroutine tgmsma(xgsm,ygsm,zgsm,xsma,ysma,zsma)

transforms_gsm_to_sma: GSM -> SM system

input : xgsm,ygsm,zgsm cartesian gsm coordinates
output: xsma,ysma,zsma cartesian sma coordinates

tsmagsm

subroutine tsmagsm(xsma,ysma,zsma,xgsm,ygsm,zgsm)

transforms_sma_to_gsm: SM -> GSM system

input : xsma,ysma,zsma cartesian sma coordinates
output: xgsm,ygsm,zgsm cartesian gsm coordinates

tsmamag

subroutine tsmamag(xsma,ysma,zsma,xmag,ymag,zmag)

transforms_sma_to_mag: SM -> MAG system

input : xsma,ysma,zsma cartesian sma coordinates
output: xmag,ymag,zmag cartesian mag coordinates

tmagsma

subroutine tmagsma(xmag,ymag,zmag,xsma,ysma,zsma)

transforms_mag_to_sma: MAG -> SM system

input : xmag,ymag,zmag cartesian mag coordinates
output: xsma,ysma,zsma cartesian sma coordinates

tgeomag

subroutine tgeomag(xgeo,ygeo,zgeo,xmag,ymag,zmag)

transforms_geo_to_mag: GEO -> MAG system

input : xgeo,ygeo,zgeo cartesian geo coordinates
output: xmag,ymag,zmag cartesian mag coordinates

tmaggeo

subroutine tmaggeo(xmag,ymag,zmag,xgeo,ygeo,zgeo)

transforms_mag_to_geo: MAG -> GEO system

input : xmag,ymag,zmag cartesian mag coordinates
output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeosma

subroutine tgeosma(xgeo,ygeo,zgeo,xsma,ysma,zsma)

transforms_geo_to_sma: GEO -> SM system

input : xgeo,ygeo,zgeo cartesian geo coordinates
output: xsma,ysma,zsma cartesian sma coordinates

tsmageo

subroutine tsmageo(xsma,ysma,zsma,xgeo,ygeo,zgeo)

transforms_sma_to_geo: SM -> GEO system

input : xsma,ysma,zsma cartesian sma coordinates
output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeogsm

subroutine tgeogsm(xgeo,ygeo,zgeo,xgsm,ygsm,zgsm)

transforms_geo_to_gsm: GEO -> GSM system

input : xgeo,ygeo,zgeo cartesian geo coordinates
output: xgsm,ygsm,zgsm cartesian gsm coordinates

tgsmgeo

subroutine tgsmgeo(xgsm,ygsm,zgsm,xgeo,ygeo,zgeo)

transforms_gsm_to_geo: GSM -> GEO system

input : xgsm,ygsm,zgsm cartesian gsm coordinates
output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeogse

subroutine tgeogse(xgeo,ygeo,zgeo,xgse,ygse,zgse)

transforms_geo_to_gse: GEO -> GSE system

input : xgeo,ygeo,zgeo cartesian geo coordinates
output: xgse,ygse,zgse cartesian gse coordinates

tgsegeo

subroutine tgsegeo(xgse,ygse,zgse,xgeo,ygeo,zgeo)

transforms_gse_to_geo: GSE -> GEO system

input : xgse,ygse,zgse cartesian gse coordinates
output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeogsq

subroutine tgeogsq(xgeo,ygeo,zgeo,xgsq,ygsq,zgsq)

transforms_geo_to_gsq: GEO -> GSEQ system

input : xgeo,ygeo,zgeo cartesian geo coordinates
output: xgsq,ygsq,zgsq cartesian gsq coordinates

tgsqgeo

subroutine tgsqgeo(xgsq,ygsq,zgsq,xgeo,ygeo,zgeo)

transforms_gsq_to_geo: GSEQ-> GEO system

input : xgsq,ygsq,zgsq cartesian gsq coordinates
output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeodme

subroutine tgeodme(xgeo,ygeo,zgeo,rlat,rlong,xdme,ydme,zdme)

transforms_geo_to_dme: GEO -> DM system

input : xgeo,ygeo,zgeo cartesian geo coordinates
 rlat,rlong latitude and longitude of the
 point of observation (radians)

output: xdme,ydme,zdme cartesian dme coordinates

tdmegeo

subroutine tdmegeo(xdme,ydme,zdme,rlat,rlong,xgeo,ygeo,zgeo)

transforms_dme_to_geo: DM -> GEO system

input : xdme,ydme,zdme cartesian dme coordinates
 rlat,rlong latitude and longitude of the
 point of observation (radians)

output: xgeo,ygeo,zgeo cartesian geo coordinates

tgeovdh

subroutine tgeovdh(xgeo,ygeo,zgeo,rlat,rlong,xvdh,yvdh,zvdh)

transforms_geo_to_vdh: GEO -> VDH system

input : xgeo,ygeo,zgeo cartesian geo coordinates
 rlat,rlong latitude and longitude of the
 point of observation (radians)

output: xvdh,yvdh,zvdh cartesian vdh coordinates

tvdhgeo

subroutine tvdhgeo(xvdh,yvdh,zvdh,rlat,rlong,xgeo,ygeo,zgeo)

transforms_vdh_to_geo: VDH -> GEO system

input : xvdh,yvdh,zvdh cartesian vdh coordinates
 rlat,rlong latitude and longitude of the
 point of observation (radians)

output: xgeo,ygeo,zgeo cartesian geo coordinates

tcarsph

subroutine tcarsph(x,y,z,r,teta,phi)

transforms_car_to_sph: CAR -> SPH system

input : x,y,z cartesian coordinates
output: r,teta,phi spherical coordinates (radians)

tsphcar

subroutine tsphcar(r,teta,phi,x,y,z)

transforms_sph_to_car: SPH -> CAR system

input : r,teta,phi spherical coordinates (radians)
output: x,y,z cartesian coordinates

6) description of "print information" subroutine

subroutine prinfo

*print_informations on the library; use gvernu
result of this call is listed in section VI*

input : none
output: print informations on terminal

VI - DIRECTION FOR USE

1) package delivered for installation

This section correspond to the file readme.txt of the delivered UNIX standard package; for DOS version, fortran source are named as *.for, and binary as *.obj; rocotexp, rocotinf and rocotche programs are detailed in the following paragraph.

ROCOTLIB Package for UNIX contents following files:

readme.txt	this file		
rocotlib.o	binary	of rocot library	
rocotexp.f	fortran code	of "example"	program
rocotexp.o	binary	of "example"	program
rocotexp.exe	executable	of "example"	program
rocotinf.f	fortran code	of "information"	program
rocotinf.o	binary	of "information"	program
rocotinf.exe	executable	of "information"	program
rocotinf.out	output	of "information"	program
rocotche.f	fortran code	of "check"	program
rocotche.o	binary	of "check"	program
rocotche.exe	executable	of "check"	program
rocotche.in	input	of "check"	program
rocotche.out	output	of "check"	program

rocotexp.exe can be used to see how use the library.

rocotinf.out contains any informations on the library;
it can be created again by:

```
rocotinf.exe >rocotinf.out
```

```
warning: original rocotinf.out is erased
```

rocotche program is a check program of the library, wich create
the output file rocotche.out from the input data in rocotche.in;
rocotche.out is created by:

```
rocotche.exe <rocotche.in
```

```
warning: original rocotche.out is erased
```

of course, it is possible to change input values into rocotche.in file
or directly answer by typing data on terminal.
obviously, rocotche.out is changed.

for new application, compilation and link can be as following:

```
f77 -c toto.f -->create toto.o  
f77 toto.o rocotlib.o -o toto.exe -->create toto.exe
```

Library has been developped on a UNIX machine with f77 compiler;
files rocotche.out correspond to results with UNIX FPS 350X machine;
a few difference can be observed on accuracy of results with another
machine.

P.Robert, november 92

2) example of Fortran user program

```
program examp

call redate(imonth,iday,iyear)
call retime(ih,im,is)

call codoty(imonth,iday,iyear,idoty)
call ctimpa(iyear,idoty,ih,im,is)
call gsundi(sxgei,sygei,szgei,sxgeo,sygeo,szgeo)
call tgeigsm(sxgei,sygei,szgei,sxgsm,sygsm,szgsm)

print*
print*, 'Sun in GEI:',sxgei,sygei,szgei
print*, 'Sun in GEO:',sxgeo,sygeo,szgeo
print*, 'Sun in GSM:',sxgsm,sygsm,szgsm

stop
end
```

Fortran code file given above, and named rocotexp.f in standard package, can be compiled with a UNIX machine as:

```
f77 -c rocotexp.f
```

which produce rocotexp.o binary file

```
f77 rocotexp.o rocotlib.o -o rocotexp.exe
```

which produce rocotexp.exe executable file
(rocotexp.o can be then erased)

both command can be grouped as:

```
f77 rocotexp.f rocotlib.o -o rocotexp.exe
```

(with no rocotexp.o produced)

and program is running with:

```
rocotexp.exe
```

which give following results:

```
rpe3d-robot 4% rocotexp.exe
imonth,iday,iyear ? (ex: 10,17,1990)
7 14 1990
hour, minute, second ? (ex: 10,45,50)
12 0 0

Sun in GEI: -0.3712461      0.8519006      0.3693803
Sun in GEO:  0.9289786      2.3603529E-02  0.3693803
Sun in GSM:  1.000000      -1.4901161E-08  0.
STOP          statement executed
rpe3d-robot 5%
```

3) output of "print information" subroutine

The small fortran program below (rocotinf.f file in standard package) give the following output containing information, in particular the version number of the library.

```
c      program rctinf
c      call prinfo
c      stop
c      end
```

will give on default output (rocotinf.out in package):

```
Output produced by call prinfo:
*****
          EUROPEAN SPACE AGENCY
          Study of the Cluster Mission
          Planning Related Aspects
          within the Numerical Simulations Network
          -----
          Coordinates Transformation Library  ROCOTLIB
          Version  1.1
          Patrick ROBERT - November 1992
*****
          AVAILABLE ROCOTLIB SUBROUTINES

1) "Read and check" subroutines
   -----
   example:
   call redatte(imonth,iday,iyear)

2) "Computation" subroutines
   -----
   example:
   call codoty(imonth,iday,iyear,idoty)

3) "Give" subroutines
   -----
   example:
   call gsundi(xgei,ygei,zgei,xgeo,ygeo,zgeo)

4) "Transformations" subroutines
   -----
   example:
   call tgeigeo(xgei,ygei,zgei,xgeo,ygeo,zgeo)

WARNING: For use of 3) and 4) subroutines,
          the ctimpa(iyear,idoty,ih,im,is) subroutine
          must be called at each time where the date
          and time are changed.
```

Example of program:

```
program examp

call redat(i,month,iday,iyear)
call retime(ih,im,is)

call codoty(i,month,iday,iyear,idoty)
call ctimpa(iyear,idoty,ih,im,is)
call gsundi(sxgei,sygei,szgei,sxgeo,sygeo,szgeo)
call tgeigsm(sxgei,sygei,szgei,sxgsm,sygsm,szgsm)

print*
print*, 'Sun in GEI:',sxgei,sygei,szgei
print*, 'Sun in GEO:',sxgeo,sygeo,szgeo
print*, 'Sun in GSM:',sxgsm,sygsm,szgsm

stop
end
```

For more information, see paper documentation

4) check program of ROCOT library

general remarks

The `rocotche.f` and `rocotche.exe` programs delivered in the standard package allows the user to check the validity of the library. It run from an input date and time, an input vector test, and an input direction of observation. Next paragraph show examples of input and output data file (`rocotche.in` and `rocotche.out` files in standard package). Accuracy of the computation is one deduce of the Sun direction computation, which is 0.006 degrees.

Input date and time allows to:

I) check calendar conversions of the library;

II) set, list and test basic time parameters as Sun direction, Dipole direction, Ecliptic pole, Sun equator pole;

III) check basic transformation, for instance verify that direction of the Sun in GSM system is well the X axis, that Dipole direction in MAG system is well the Z axis, and so on;

Input vector test allows to:

IV) check the transformations given in the library, by changing the input vector across all available paths showed on the schematic diagram of § II-2.

Mainly, the vector is transformed:

a) as a "star" centred on GEI system (see schematic diagram);

b) as a "counter-clockwise ring cumulative transformation", from GEO to GEO after a complete circle;

c) as a "clockwise ring cumulative transformation", from GEO to GEO, in the other sense;

d) as a "star" centred on GEO system;

e) Finally, input direction of the point of observation is used to check local coordinate systems, i.e. DM and VDH.

At each "go and back", one check that the original vector is returned to the same value, taking into account accuracy of the computation (32 bits simple precision, 7 significative digits, and input Sun direction within 0.006 degrees).

input data file of check program

(file rocotche.in in the standard package)

```
10 17 1990
12 30 1
10. 30. 60.
45. 30.

-----
this file is a file example of input data for rocotche program

input:

imonth, iday, iyear
ihour, imin, isec
R, teta, phi of anyone test vector (teta, phi in degrees)
geographic lat. and long. of point of observation for local systems test

-----
```

output result of check program

(file rocotche.out in the standard package)

```
          T E S T   O F   R O C O T L I B   -   V e r s i o n   1 . 1

I) TEST OF ROCOTLIB CALENDAR CONVERSIONS
-----

INPUT DATE AND TIME:

month, day, year:           10    17  1990
hour, minute, second:      12    30    1

computed values:

decimal hour:                12.50028
day of the year:             290
Julian day from 1-1-1950:   14900
leap year (1=yes,0=no):     0

recompute month and day from year and day of the year:    10    17
recompute date from Julian day :    10    17  1990
recompute time from decimal hour:    12    30    1
```

suite --->

II) CHECK BASIC TIME PARAMETERS

SUN PARAMETERS IN GEI SYSTEM

Greenwich Sideral Time (deg.): 213.25262
ecliptic longitude (deg.): 203.87914
right ascension (deg.): 202.10098
declination (deg.): -9.26467

DIPOLE TILT ANGLE (deg.): -3.73804

SUN DIRECTION

in GEI:	X= -0.9144359E+00	r = 0.1000000E+01
	Y= -0.3713321E+00	teta= 99.26467 (deg.)
	Z= -0.1609952E+00	phi = -157.89903 (deg.)
in GEO:	X= 0.9683203E+00	r = 0.1000000E+01
	Y= -0.1908833E+00	teta= 99.26467 (deg.)
	Z= -0.1609952E+00	phi = -11.15164 (deg.)

DIPOLE DIRECTION

in GEI:	X= -0.1484070E+00	r = 0.1000000E+01
	Y= 0.1148030E+00	teta= 10.81440 (deg.)
	Z= 0.9822402E+00	phi = 142.27560 (deg.)
in GEO:	X= 0.6115692E-01	r = 0.1000000E+01
	Y= -0.1773815E+00	teta= 10.81440 (deg.)
	Z= 0.9822402E+00	phi = -70.97702 (deg.)

ECLIPTIC DIRECTION

in GEI:	X= 0.0000000E+00	r = 0.1000000E+01
	Y= -0.3979853E+00	teta= 23.45229 (deg.)
	Z= 0.9173918E+00	phi = -90.00000 (deg.)
in GEO:	X= 0.2182279E+00	r = 0.1000000E+01
	Y= 0.3328196E+00	teta= 23.45229 (deg.)
	Z= 0.9173918E+00	phi = 56.74738 (deg.)

SUN EQUATOR DIRECTION

in GEI:	X= 0.1218259E+00	r = 0.1000000E+01
	Y= -0.4233948E+00	teta= 26.14045 (deg.)
	Z= 0.8977168E+00	phi = -73.94758 (deg.)
in GEO:	X= 0.1302824E+00	r = 0.1000000E+01
	Y= 0.4208696E+00	teta= 26.14045 (deg.)
	Z= 0.8977168E+00	phi = 72.79982 (deg.)

suite --->

III) CHECK BASIC TRANSFORMATIONS

Check (S,E) angle is equal to 90 deg. : angle= 89.9949 (deg.)

1) input vector: Sun direction in GEI system

X= -0.9144359E+00 r = 0.1000000E+01
Y= -0.3713321E+00 teta= 99.26467 (deg.)
Z= -0.1609952E+00 phi = -157.89903 (deg.)

in GEO system

X= 0.9683203E+00 r = 0.1000000E+01
Y= -0.1908833E+00 teta= 99.26467 (deg.)
Z= -0.1609952E+00 phi = -11.15164 (deg.)

check Y=0 in SM system with tgeisma:

X= 0.9978815E+00 r = 0.1000000E+01
Y= 0.2421439E-07 teta= 93.73015 (deg.)
Z= -0.6505735E-01 phi = 0.00000 (deg.)

with tgeosma:

X= 0.9978816E+00 r = 0.1000000E+01
Y= 0.9313226E-08 teta= 93.73015 (deg.)
Z= -0.6505734E-01 phi = 0.00000 (deg.)

check Y=0 and Z=0 in GSM system with tgeigsm:

X= 0.1000000E+01 r = 0.1000000E+01
Y= 0.2421439E-07 teta= 90.00000 (deg.)
Z= 0.0000000E+00 phi = 0.00000 (deg.)

with tgeogsm:

X= 0.1000000E+01 r = 0.1000000E+01
Y= 0.9313226E-08 teta= 90.00000 (deg.)
Z= 0.1490116E-07 phi = 0.00000 (deg.)

check Y=0 and Z=0 in GSE system with tgeigse:

X= 0.1000000E+01 r = 0.1000000E+01
Y= -0.7450581E-08 teta= 89.99490 (deg.)
Z= 0.8901954E-04 phi = 0.00000 (deg.)

with tgeogse:

X= 0.1000000E+01 r = 0.1000000E+01
Y= 0.0000000E+00 teta= 89.99490 (deg.)
Z= 0.8901954E-04 phi = 0.00000 (deg.)

check Y=0 and Z=0 in GSQ system with tgeigsq:

X= 0.1000000E+01 r = 0.1000000E+01
Y= -0.2980232E-07 teta= 90.00000 (deg.)
Z= 0.1490116E-07 phi = 0.00000 (deg.)

with tgeogsq:

X= 0.1000000E+01 r = 0.1000000E+01
Y= -0.7450581E-08 teta= 90.00000 (deg.)
Z= 0.0000000E+00 phi = 0.00000 (deg.)

suite --->

2) input vector: Dipole direction in GEI system

X= -0.1484070E+00 r = 0.1000000E+01
Y= 0.1148030E+00 teta= 10.81440 (deg.)
Z= 0.9822402E+00 phi = 142.27560 (deg.)

in GEO system

X= 0.6115692E-01 r = 0.1000000E+01
Y= -0.1773815E+00 teta= 10.81440 (deg.)
Z= 0.9822402E+00 phi = -70.97702 (deg.)

check X=0 and Y=0 in MAG system with tgeimag:

X= -0.1490116E-07 r = 0.1000000E+01
Y= 0.0000000E+00 teta= 0.00000 (deg.)
Z= 0.1000000E+01 phi = 180.00001 (deg.)

with tgeomag:

X= -0.1490116E-07 r = 0.1000000E+01
Y= 0.0000000E+00 teta= 0.00000 (deg.)
Z= 0.1000000E+01 phi = 180.00001 (deg.)

check X=0 and Y=0 in SM system with tgeisma:

X= -0.1490116E-07 r = 0.1000000E+01
Y= -0.1490116E-07 teta= 0.00000 (deg.)
Z= 0.1000000E+01 phi = -135.00000 (deg.)

with tgeosma:

X= 0.0000000E+00 r = 0.1000000E+01
Y= -0.1490116E-07 teta= 0.00000 (deg.)
Z= 0.1000000E+01 phi = -90.00000 (deg.)

check Y=0 in GSM system with tgeigsm:

X= -0.6505735E-01 r = 0.1000000E+01
Y= -0.1490116E-07 teta= 3.73015 (deg.)
Z= 0.9978816E+00 phi = -179.99999 (deg.)

with tgeogsm:

X= -0.6505734E-01 r = 0.1000000E+01
Y= -0.1490116E-07 teta= 3.73015 (deg.)
Z= 0.9978816E+00 phi = -179.99999 (deg.)

suite --->

3) input vector: Ecliptic direction in GEI system

X= 0.0000000E+00 r = 0.1000000E+01
Y= -0.3979853E+00 teta= 23.45229 (deg.)
Z= 0.9173918E+00 phi = -90.00000 (deg.)

in GEO system

X= 0.2182279E+00 r = 0.1000000E+01
Y= 0.3328196E+00 teta= 23.45229 (deg.)
Z= 0.9173918E+00 phi = 56.74738 (deg.)

check X=0 and Y=0 in GSE system with tgeigse:

X= 0.8901954E-04 r = 0.1000000E+01
Y= 0.0000000E+00 teta= 0.00510 (deg.)
Z= 0.1000000E+01 phi = 0.00000 (deg.)

with tgeogse:

X= 0.8901954E-04 r = 0.1000000E+01
Y= 0.0000000E+00 teta= 0.00510 (deg.)
Z= 0.1000000E+01 phi = 0.00000 (deg.)

check X=0 in GSQ system with tgeigsq:

X= 0.8901954E-04 r = 0.9999999E+00
Y= -0.7816470E-01 teta= 4.48308 (deg.)
Z= 0.9969404E+00 phi = -89.93475 (deg.)

with tgeogsq:

X= 0.8901954E-04 r = 0.1000000E+01
Y= -0.7816467E-01 teta= 4.48308 (deg.)
Z= 0.9969406E+00 phi = -89.93475 (deg.)

4) input vector: Sun equator in GEI system

X= 0.1218259E+00 r = 0.1000000E+01
Y= -0.4233948E+00 teta= 26.14045 (deg.)
Z= 0.8977168E+00 phi = -73.94758 (deg.)

in GEO system

X= 0.1302824E+00 r = 0.1000000E+01
Y= 0.4208696E+00 teta= 26.14045 (deg.)
Z= 0.8977168E+00 phi = 72.79982 (deg.)

check and Y=0 in GSQ system with tgeigsq:

X= -0.9871002E-01 r = 0.1000000E+01
Y= -0.2980232E-07 teta= 5.66489 (deg.)
Z= 0.9951164E+00 phi = -179.99999 (deg.)

with tgeogsq:

X= -0.9871002E-01 r = 0.1000000E+01
Y= 0.0000000E+00 teta= 5.66489 (deg.)
Z= 0.9951165E+00 phi = 180.00001 (deg.)

suite --->

5) input vector: North geographic in GEO system

X=	0.0000000E+00	r	=	0.1000000E+01
Y=	0.0000000E+00	teta=		0.00000 (deg.)
Z=	0.1000000E+01	phi =		0.00000 (deg.)

check X=0 and Y=0 in GEI system with tgeogei:

X=	0.0000000E+00	r	=	0.1000000E+01
Y=	0.0000000E+00	teta=		0.00000 (deg.)
Z=	0.1000000E+01	phi =		0.00000 (deg.)

check Y=0 in MAG system with tgeimag:

X=	-0.1876283E+00	r	=	0.1000000E+01
Y=	0.0000000E+00	teta=		10.81440 (deg.)
Z=	0.9822402E+00	phi =		180.00001 (deg.)

with tgeomag:

X=	-0.1876283E+00	r	=	0.1000000E+01
Y=	0.0000000E+00	teta=		10.81440 (deg.)
Z=	0.9822402E+00	phi =		180.00001 (deg.)

IV) TEST OF ROCOTLIB TRANSFORMATION SUBROUTINES

1) input choosed test vector in GEO system :

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330127E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

2) converts GEO system to GEI system by tgeogei

X=	0.2836924E+00	r	=	0.1000000E+02
Y=	-0.4991945E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		-86.74738 (deg.)

suite --->

a) STAR TRANSFORMATIONS AROUND GEI SYSTEM

3) converts GEI system to MAG system by tgeimag

X=	-0.4845459E+01	r	=	0.1000000E+02
Y=	0.3774863E+01	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		142.07957 (deg.)

come back to GEI system by tmaggei

X=	0.2836928E+00	r	=	0.1000000E+02
Y=	-0.4991945E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		-86.74738 (deg.)

*** difference between first GEI vector: angle= 0.0000 (deg.)
ratio= 0.9999999E+00

4) converts GEI system to SM system by tgeisma

X=	0.7148904E+00	r	=	0.9999999E+01
Y=	0.6100573E+01	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		83.31633 (deg.)

come back to GEI system by tsmagei

X=	0.2836916E+00	r	=	0.1000000E+02
Y=	-0.4991944E+01	teta=		29.99999 (deg.)
Z=	0.8660254E+01	phi =		-86.74739 (deg.)

*** difference between first GEI vector: angle= 0.0000 (deg.)
ratio= 0.9999999E+00

5) converts GEI system to GSM system by tgeigsm

X=	0.1999916E+00	r	=	0.9999999E+01
Y=	0.6100573E+01	teta=		37.61735 (deg.)
Z=	0.7921048E+01	phi =		88.12238 (deg.)

come back to GEI system by tgsngei

X=	0.2836920E+00	r	=	0.9999999E+01
Y=	-0.4991944E+01	teta=		29.99999 (deg.)
Z=	0.8660254E+01	phi =		-86.74738 (deg.)

*** difference between first GEI vector: angle= 0.0000 (deg.)
ratio= 0.1000000E+01

6) converts GEI system to GSE system by tgeigse

X=	0.1999916E+00	r	=	0.1000002E+02
Y=	0.1150798E+01	teta=		6.70771 (deg.)
Z=	0.9931566E+01	phi =		80.14131 (deg.)

come back to GEI system by tgsegei

X=	0.2828839E+00	r	=	0.1000003E+02
Y=	-0.4992280E+01	teta=		30.00179 (deg.)
Z=	0.8660128E+01	phi =		-86.75684 (deg.)

*** difference between first GEI vector: angle= 0.0051 (deg.)
ratio= 0.9999965E+00

suite --->

b) COUNTER-CLOCKWISE RING CUMULATIVE TRANSFORMATIONS

input choosed test vector in GEO system :

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330127E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

2) converts GEO system to GEI system by tgeogei

X=	0.2836924E+00	r	=	0.1000000E+02
Y=	-0.4991945E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		-86.74738 (deg.)

7) converts GEI system to GSQ system by tgeigsq

X=	0.1999916E+00	r	=	0.1000000E+02
Y=	0.3709805E+00	teta=		2.41547 (deg.)
Z=	0.9991115E+01	phi =		61.67138 (deg.)

8) converts GSQ system to GSE system by tgsqgse

X=	0.1999916E+00	r	=	0.1000000E+02
Y=	0.1150798E+01	teta=		6.70773 (deg.)
Z=	0.9931549E+01	phi =		80.14131 (deg.)

9) converts GSE system to GSM system by tgsegsm

X=	0.1999916E+00	r	=	0.9978775E+01
Y=	0.6087644E+01	teta=		37.61759 (deg.)
Z=	0.7904211E+01	phi =		88.11839 (deg.)

10) converts GSM system to SM system by tgsmsma

X=	0.7133721E+00	r	=	0.9965565E+01
Y=	0.6087644E+01	teta=		37.95540 (deg.)
Z=	0.7857746E+01	phi =		83.31636 (deg.)

11) converts SM system to MAG system by tsmamag

X=	-0.4835190E+01	r	=	0.9965564E+01
Y=	0.3766860E+01	teta=		37.95540 (deg.)
Z=	0.7857746E+01	phi =		142.07960 (deg.)

12) converts MAG system to GEO system by tmaggeo

X=	0.2493672E+01	r	=	0.9965565E+01
Y=	0.4323928E+01	teta=		30.05768 (deg.)
Z=	0.8625412E+01	phi =		60.02734 (deg.)

*** difference between first GEO vector: angle= 0.0593 (deg.)
ratio= 0.1003455E+01

suite --->

c) CLOCKWISE RING CUMULATIVE TRANSFORMATIONS

input choosed test vector in GEO system :

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330127E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

-12) converts GEO system to MAG system by tgeomag

X=	-0.4845460E+01	r	=	0.1000000E+02
Y=	0.3774863E+01	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		142.07957 (deg.)

-11) converts MAG system to SM system by tmagsma

X=	0.7148898E+00	r	=	0.1000000E+02
Y=	0.6100574E+01	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		83.31633 (deg.)

-10) converts SM system to GSM system by tsmagsm

X=	0.1984798E+00	r	=	0.9986761E+01
Y=	0.6100574E+01	teta=		37.67555 (deg.)
Z=	0.7904366E+01	phi =		88.13656 (deg.)

-9) converts GSM system to GSE system by tgsmsgse

X=	0.1984798E+00	r	=	0.9965565E+01
Y=	0.1156897E+01	teta=		6.76432 (deg.)
Z=	0.9896194E+01	phi =		80.26499 (deg.)

-8) converts GSE system to GSQ system by tgsegsq

X=	0.1984798E+00	r	=	0.9965564E+01
Y=	0.3798242E+00	teta=		2.46469 (deg.)
Z=	0.9956345E+01	phi =		62.41038 (deg.)

-7) converts GSQ system to GEI system by tgsqgei

X=	0.2875404E+00	r	=	0.9965563E+01
Y=	-0.4982431E+01	teta=		30.05273 (deg.)
Z=	0.8625842E+01	phi =		-86.69708 (deg.)

*** difference between first GEI vector: angle= 0.0584 (deg.)
ratio= 0.1003456E+01

-2) converts GEI system to GEO system by tgeigeo

X=	0.2491565E+01	r	=	0.9965563E+01
Y=	0.4324281E+01	teta=		30.05273 (deg.)
Z=	0.8625842E+01	phi =		60.05031 (deg.)

*** difference between first GEO vector: angle= 0.0584 (deg.)
ratio= 0.1003456E+01

suite --->

d) STAR TRANSFORMATIONS AROUND GEO SYSTEM

input choosed test vector in GEO system :

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330127E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

13) converts GEO system to SM system by tgeosma

X=	0.7148904E+00	r	=	0.9999999E+01
Y=	0.6100573E+01	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		83.31633 (deg.)

come back to GEO system by tsmageo

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330126E+01	teta=		29.99999 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.1000000E+01

14) converts GEO system to GSM system by tgeogsm

X=	0.1999917E+00	r	=	0.9999999E+01
Y=	0.6100573E+01	teta=		37.61735 (deg.)
Z=	0.7921048E+01	phi =		88.12238 (deg.)

come back to GEO system by tgsageo

X=	0.2500000E+01	r	=	0.9999999E+01
Y=	0.4330126E+01	teta=		29.99999 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.1000000E+01

15) converts GEO system to GSE system by tgeogse

X=	0.1999917E+00	r	=	0.1000002E+02
Y=	0.1150798E+01	teta=		6.70771 (deg.)
Z=	0.9931566E+01	phi =		80.14130 (deg.)

come back to GEO system by tgsegeo

X=	0.2500860E+01	r	=	0.1000003E+02
Y=	0.4329965E+01	teta=		30.00180 (deg.)
Z=	0.8660128E+01	phi =		59.99054 (deg.)

*** difference between first GEO vector: angle= 0.0051 (deg.)
ratio= 0.9999964E+00

16) converts GEO system to GSQ system by tgeogsq

X=	0.1999917E+00	r	=	0.1000000E+02
Y=	0.3709810E+00	teta=		2.41547 (deg.)
Z=	0.9991116E+01	phi =		61.67139 (deg.)

come back to GEO system by tgsqgeo

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330128E+01	teta=		30.00000 (deg.)
Z=	0.8660255E+01	phi =		60.00001 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.9999998E+00

e) LOCAL SYSTEMS

with direction of observation=input GEO vector
geog. lat., long. = 60.00000 60.00000

17) converts GEO system to DM system by tgeodme

X=	0.6142318E+01	r	=	0.1000000E+02
Y=	-0.2384186E-06	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		0.00000 (deg.)

come back to GEO system by tdmegeo

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330128E+01	teta=		30.00001 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.9999999E+00

18) converts GEO system to VDH system by tgeovdh

X=	0.1000000E+02	r	=	0.1000000E+02
Y=	0.0000000E+00	teta=		90.00001 (deg.)
Z=	-0.1430511E-05	phi =		0.00000 (deg.)

come back to GEO system by tvdhgeo

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330128E+01	teta=		30.00001 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.9999999E+00

with anyone choosed direction of observation
geog. lat., long. = 45.00000 30.00000

17-b) converts GEO system to DM system by tgeodme

X=	0.5254447E+01	r	=	0.1000000E+02
Y=	0.3181014E+01	teta=		37.89612 (deg.)
Z=	0.7891257E+01	phi =		31.19049 (deg.)

come back to GEO system by tdmegeo

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330128E+01	teta=		30.00001 (deg.)
Z=	0.8660254E+01	phi =		60.00000 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.9999999E+00

18-b) converts GEO system to VDH system by tgeovdh

X=	0.9185587E+01	r	=	0.1000000E+02
Y=	0.2500000E+01	teta=		72.17046 (deg.)
Z=	0.3061862E+01	phi =		15.22516 (deg.)

come back to GEO system by tvdhgeo

X=	0.2500000E+01	r	=	0.1000000E+02
Y=	0.4330128E+01	teta=		30.00000 (deg.)
Z=	0.8660254E+01	phi =		60.00001 (deg.)

*** difference between first GEO vector: angle= 0.0000 (deg.)
ratio= 0.9999999E+00

end of test program

5) Summary of available subroutine

"read and check" subroutines:

long name	call	arguments	comments
read_date	redate	(imonth, iday, iyear)	& check validity
read_time	retime	(ih, im, is)	& check validity

"computation" subroutines:

long name	call	arguments	comments
compute_time_parameters	ctimpa	(iyear, idoty, ihour, imin, isec)	prepare matrix
compute_dipole_direction	cdipdi	(iyear, idoty, d1, d2, d3)	in GEO system
compute_sun_direction	csundi	(iyear, idoty, ih, im, is, gst, slon, sra, sdec)	in GEI system
compute_leap_year	cleapy	(iyear, ileap)	ileap=1:leap year
compute_day_of_the_year	codoty	(imonth, iday, iyear, idoty)	idoty=1:January 1
compute_Julian_day	cjud50	(imonth, iday, iyear, jud50)	jud50=1:1,1,1950
compute_date_from_day_of_the_year	cdfdot	(iyear, idoty, imonth, iday)	and a given year
compute_date_from_Julian_day	cdfjud	(jud50, imonth, iday, iyear)	jud50=1:1,1,1950
compute_decimal_hour	codech	(ih, im, is, dech)	from ih, im, is
compute_time	cotime	(dech, ih, im, is)	from decimal hour
compute_angle_and_ratio	canara	(ux, uy, uz, vx, vy, vz, angle, ratio)	between U and V

"give" subroutines:

long name	call	arguments	comments
give_sun_direction	gsundi	(sxgei, sygei, szgei, sxgeo, sygeo, szgeo)	in GEI and GEO
give_dipole_direction	gdipdi	(dxgei, dygei, dzgei, dxgeo, dygeo, dzgeo)	in GEI and GEO
give_ecliptic_direction	gecldi	(exgei, eygei, ezgei, exgeo, eygeo, ezgeo)	in GEI and GEO
give_sun_rotation_direction	gsrodi	(rxgei, rygei, rzgei, rxgeo, rygeo, rzgeo)	in GEI and GEO
give_sun_parameter	gsunpa	(gmst, slon, sras, sdec)	in GEI (rad.)
give_dipole_tilt_angle	gdipta	(dipta)	in GSM (rad.)
give_version_number	gvernu	(vernu)	of the library

"transformation" subroutines:

long name	call	arguments	comments
transforms_gei_to_geo	tgeigeo	(xgei,ygei,zgei,xgeo,ygeo,zgeo)	GEI -> GEO
transforms_geo_to_gei	tgeogei	(xgeo,ygeo,zgeo,xgei,ygei,zgei)	GEO -> GEI
transforms_gei_to_mag	tgeimag	(xgei,ygei,zgei,xmag,ymag,zmag)	GEI -> MAG
transforms_mag_to_gei	tmaggei	(xmag,ymag,zmag,xgei,ygei,zgei)	MAG -> GEI
transforms_gei_to_sma	tgeisma	(xgei,ygei,zgei,xsma,ysma,zsma)	GEI -> SM
transforms_sma_to_gei	tsmagei	(xsma,ysma,zsma,xgei,ygei,zgei)	SM -> GEI
transforms_gei_to_gsm	tgeigsm	(xgei,ygei,zgei,xgsm,ygsm,zgsm)	GEI -> GSM
transforms_gsm_to_gei	tgsmgei	(xgsm,ygsm,zgsm,xgei,ygei,zgei)	GSM -> GEI
transforms_gei_to_gse	tgeigse	(xgei,ygei,zgei,xgse,ygse,zgse)	GEI -> GSE
transforms_gse_to_gei	tgsegei	(xgse,ygse,zgse,xgei,ygei,zgei)	GSE -> GEI
transforms_gei_to_gsq	tgeigsq	(xgei,ygei,zgei,xgsq,ygsq,zgsq)	GEI -> GSEQ
transforms_gsq_to_gei	tgsqgei	(xgsq,ygsq,zgsq,xgei,ygei,zgei)	GSEQ-> GEI
transforms_gse_to_gsq	tgsegsq	(xgse,ygse,zgse,xgsq,ygsq,zgsq)	GSE -> GSEQ
transforms_gsq_to_gse	tgsqgse	(xgsq,ygsq,zgsq,xgse,ygse,zgse)	GSEQ-> GSE
transforms_gse_to_gsm	tgsegsm	(xgse,ygse,zgse,xgsm,ygsm,zgsm)	GSE -> GSM
transforms_gsm_to_gse	tgsmgse	(xgsm,ygsm,zgsm,xgse,ygse,zgse)	GSM -> GSE
transforms_gsm_to_sma	tgsmsma	(xgsm,ygsm,zgsm,xsma,ysma,zsma)	GSM -> SM
transforms_sma_to_gsm	tsmagsm	(xsma,ysma,zsma,xgsm,ygsm,zgsm)	SM -> GSM
transforms_sma_to_mag	tsmamag	(xsma,ysma,zsma,xmag,ymag,zmag)	SM -> MAG
transforms_mag_to_sma	tmagsma	(xmag,ymag,zmag,xsma,ysma,zsma)	MAG -> SM
transforms_geo_to_mag	tgeomag	(xgeo,ygeo,zgeo,xmag,ymag,zmag)	GEO -> MAG
transforms_mag_to_geo	tmaggeo	(xmag,ymag,zmag,xgeo,ygeo,zgeo)	MAG -> GEO
transforms_geo_to_sma	tgeosma	(xgeo,ygeo,zgeo,xsma,ysma,zsma)	GEO -> SM
transforms_sma_to_geo	tsmageo	(xsma,ysma,zsma,xgeo,ygeo,zgeo)	SM -> GEO
transforms_geo_to_gsm	tgeogsm	(xgeo,ygeo,zgeo,xgsm,ygsm,zgsm)	GEO -> GSM
transforms_gsm_to_geo	tgsmgeo	(xgsm,ygsm,zgsm,xgeo,ygeo,zgeo)	GSM -> GEO
transforms_geo_to_gse	tgeogse	(xgeo,ygeo,zgeo,xgse,ygse,zgse)	GEO -> GSE
transforms_gse_to_geo	tgsegeo	(xgse,ygse,zgse,xgeo,ygeo,zgeo)	GSE -> GEO
transforms_geo_to_gsq	tgeogsq	(xgeo,ygeo,zgeo,xgsq,ygsq,zgsq)	GEO -> GSEQ
transforms_gsq_to_geo	tgsqgeo	(xgsq,ygsq,zgsq,xgeo,ygeo,zgeo)	GSEQ-> GEO
transforms_geo_to_dme	tgeodme	(xgeo,ygeo,zgeo,rlat,rlong,xdme,ydme,zdme)	GEO -> DM
transforms_dme_to_geo	tdmegeo	(xdme,ydme,zdme,rlat,rlong,xgeo,ygeo,zgeo)	DM -> GEO
transforms_geo_to_vdh	tgeovdh	(xgeo,ygeo,zgeo,rlat,rlong,xvdh,yvdh,zvdh)	GEO -> VDH
transforms_vdh_to_geo	tvdhgeo	(xvdh,yvdh,zvdh,rlat,rlong,xgeo,ygeo,zgeo)	VDH -> GEO
transforms_car_to_sph	tcarsph	(x,y,z,r,teta,phi)	CAR -> SPH
transforms_sph_to_car	tsphcar	(r,teta,phi,x,y,z)	SPH -> CAR

"print information" subroutine

long name	call	arguments	comments
print_informations	prinfo	none	

Bibliography

Geophysical coordinate transformations, C.T. Russell, cosmic electrodynamics, v.2, 184-196, 1971.

NOTES

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